

Jonathan Ebersole  
235 E. Farimount Ave. Apt. 102  
State College, PA 16801

September 25, 2013  
Dr. Linda Hanagan  
Associate Professor  
212 Engineering Unit A  
University Park, Pa 16802

Dear Dr. Hanagan,

This document is developed to help guide you through the evaluation of a typical bay under gravity loads and the evaluation of three alternative framing systems for the Oklahoma University Children's Medical Office Building. The purpose of this assignment is to evaluate a typical bay under gravity loads and to analyze three different structural systems that I could use in my proposal. The document contains a site plan of the building along with a list of codes and documents used to determine the member sizes. The calculations contain the loading, existing member analysis, and three different alternative systems. The calculations are accompanied by sketches of existing and proposed bays. For each of the four systems, I analyzed the flexural strength, the shear strength, and the deflections of each member. A column evaluation was conducted for the existing system. For my three alternative systems, I chose to use non-composite steel, composite steel, and a one way slab with beams.

Sincerely,

Jonathan Ebersole

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# OU Children's Medical Office Building

Jonathan Ebersole | Structural Option

<http://www.engr.psu.edu/ae/thesis/portfolios/2014/jme5193/index.html>



## Project Team

- Owner: University Hospitals Trust
- Construction Manager: Flintco, Inc.
- Project Architect: Miles Associates
- Design Architect: Hellmuth, Obata, and Kassabaum, Inc.
- Structural Engineer: Zahl-Ford
- MEP Engineer: ZRHD, P.C.
- Civil Engineer: Smith, Roberts, Baldischwiler, Inc.

## General Information

- Location: 1200 North Children's Avenue, Oklahoma City, Oklahoma
- Occupancy: Office
- Size: 320,000 sq. ft.
- Height: 12 Stories for a total of 172 ft.
- Construction Dates: February 2007-Spring of 2009
- Building Cost: \$59,760,000
- Delivery Method: Design-Bid-Build

## Architecture

- Exterior Façade comprised of brick Veneer with large glass curtain wall on the front face of the building
- Supports Hospital with additional office space, exam rooms, and labs
- Membrane roof system with rigid insulation and light weight insulating concrete

## Structural Design

- Reinforced concrete columns and beams
- 10" thick flat slab system with drop panels
- Concrete shear walls located in elevator shafts and stairwells
- Drilled pier foundation with a minimum bearing capacity of 45 KSF

## Mechanical Design

- 7,500 CFM Air Handling unit occupies each floor
- Heat Exchanger is used to heat water before entering the heating coil
- 850 CFM fans are used to pressurize the stairwells

## Lighting/Electrical Design

- Service voltage is 480/277 V, three phase, with 4 wires
- Voltage reduced to 120/208V, three phase, with 4 wires and supplied to each panel box
- Fluorescent lamps are used throughout the building to save energy costs

## **General Information**

### **Executive Summary**

OU Children's Medical Office Building is an office building located in Oklahoma. It is situated next to an existing hospital and parking garage. The building houses offices, examination rooms and labs for the expanding OU Children's Hospital. It is the largest free standing clinical office in the state and provides much needed medical services to the children of Oklahoma and their families.

The structure of the building is reinforced concrete. The building uses a flat slab system supported by columns and exterior beams. Drop panels are used at the column locations to provide extra shear and moment capacity to the slab. The columns are supported on piers that transfer the loads to bedrock underneath the building. The building also uses shear walls and moment frames to resist the lateral forces.

This building provides several unique challenges that a typical office building would not otherwise have. These include a parking garage located on the first floor, a future helicopter pad positioned on the roof, and impact loads on lower levels for vehicle collisions with the building. These design parameters will increase the difficulty of future design assignments as all load cases must be analyzed.

## Site Plan

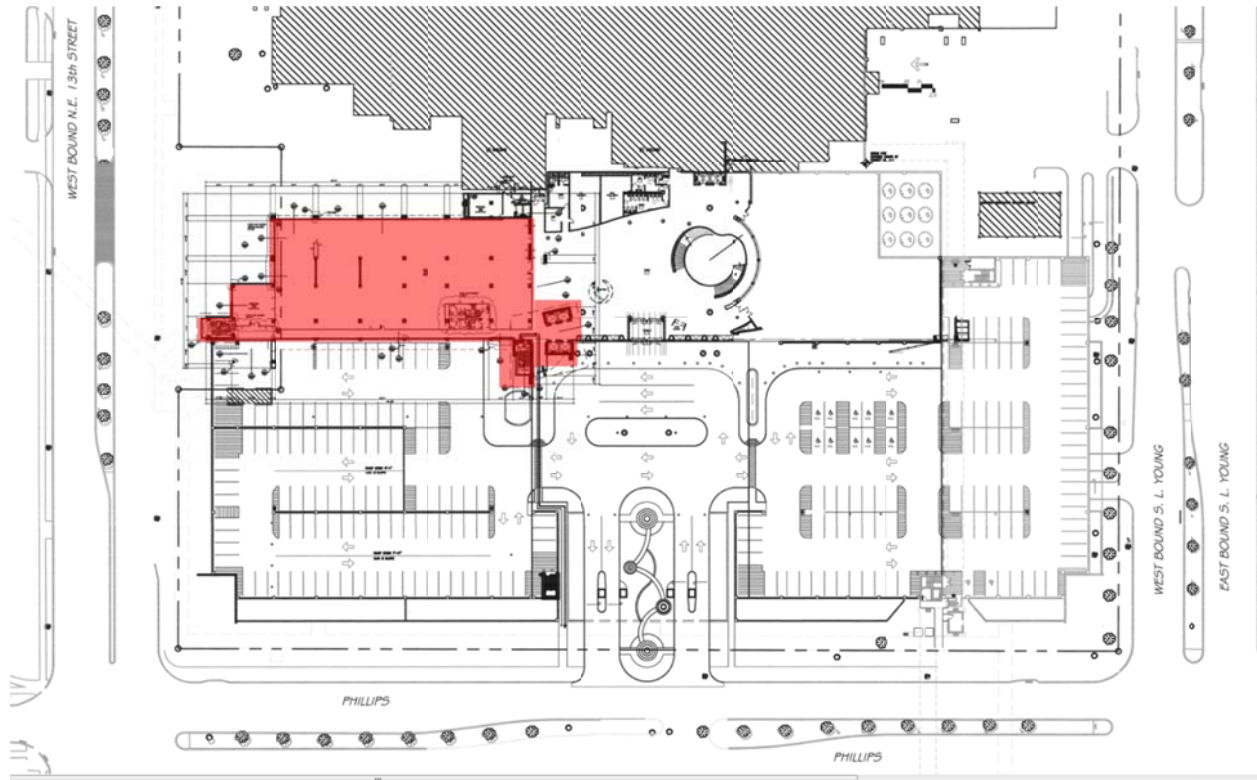


Figure A. Building outlined in red.

OU Children's Medical Office Building is located on 1200 N. Children's Avenue Oklahoma City, Oklahoma between Stanton L. Young Blvd and N.E. 13th Street. (Refer to figure A for site and building footprint). The building is twelve stories above grade and is approximately 180 feet tall. The building footprint is 22,820 square feet with a total area of 320,000 square feet. The building is positioned between an existing hospital and existing parking structure. A large atrium connects the hospital to the office building and parking structure but it is a future addition and not part of the original office construction. The building is located in an urbanized area which will later impact the design for the lateral loads.

## List of Documents

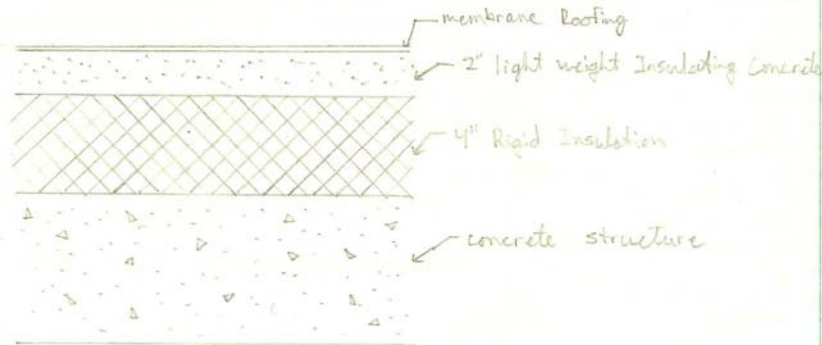
For this assignment, several documents were used in order to evaluate and design the required members. The ACI 318-02 code was used to analyze the existing structure, whereas, the ACI 318-11 was used to design the one way slab and beams in one of my

alternative systems. I also used examples and design aids from the sixth edition of *Reinforced Concrete Mechanics and Design* written by James Wight and James MacGregor. To design the composite decking found in two of my alternative systems, I used the Vulcraft deck catalog. To determine the member sizes for these two systems, the fourteenth edition of the AISC Steel Construction Manual was used. Notes from AE 401, AE 402, AE 403 and AE 431 were also used.

2-1

1. Gravity Loads

1.a. Roof



cross section of Typical Roof Construction

Loads

Live Load - 20 psf (IBC 2003)

Dead Load

Materials

Membrane Roof	ASCE 7-02	- 1 psf
Light Weight Insulating Concrete (2" thick)	ASCE 7-02 $15 \text{pcf} \cdot \frac{2 \text{in}}{12 \text{in}}$	- 19 psf
Rigid Insulation (4" thick)	AISC 14 <sup>th</sup> ed. $15 \text{psf/in} \cdot 4 \text{in}$	- 6 psf
Superimposed Dead Load		- 15 psf

Structure

slabs (10" thick)	$150 \text{pcf} \cdot \frac{10 \text{in}}{12 \text{in}}$	- 125 psf
drop panels (4" thick)	$150 \text{pcf} \cdot \frac{4 \text{in}}{12 \text{in}} = \frac{10 \cdot 67 \cdot 7.33}{24 \cdot 32}$	- 5 psf
column (22" x 22")	$150 \text{pcf} \cdot \frac{(14 \text{in} \cdot 14 \text{in}) \cdot 22 \text{in} \cdot 22 \text{in}}{12 \text{in} \cdot 12 \text{in} \cdot 24 \cdot 32}$	- 7 psf

2-2

## Snow Loads

Flat Roof Snow Loads (ASCE 7-02)  
 slope =  $1/4''$  per foot  $\approx 5^\circ$  slope  $\rightarrow$  fits criteria

$$p_f = 0.7 C_e C_t I p_g$$

$$C_e = 1.0$$

$$C_t = 1.1$$

$$I = 1.0$$

$$p_g = 10 \text{ psf}$$

$$p_f = 0.7 \cdot 1.0 \cdot 1.1 \cdot 1.0 \cdot 10 \text{ psf} = 7.7 \text{ psf}$$

but not less than:

$p_g$  is 20 psf or less

$$p_f = I(p_g) = 1.0 \cdot 10 = 10 \text{ psf} \rightarrow \text{controls}$$

## Snow Drift Loads

## Parapet

$$l_w = 91.29 \text{ ft}$$

$$h_d = 2.5 \text{ ft}$$

$$h_c = 4.683 \text{ ft}$$

$$\lambda = 0.13 p_g h_c = 0.13 \cdot 10 \cdot 4.683 = 15.3 \text{ psf}$$

$$h_w = p_f / \lambda = 10 / 15.3 = 0.65 \text{ ft}$$

$$w = 4 h_d = 4 \cdot 2.5 = 10 \text{ ft}$$

$$h_c / h_w = 4.683 / 0.65 = 7.2 > 0.2 \rightarrow \text{must be applied}$$

$$p_d = 3/4 h_d \lambda = 3/4 \cdot 2.5 \cdot 15.3 = 28.69 \text{ psf (windward)}$$

$$p_d \cdot 0.5 \cdot w = 28.69 \cdot 0.5 \cdot 10 = 143.45 \text{ plf}$$

## Load combinations

$$1/2 D + 0.5 L_r \text{ or } S$$

$$1/2 D + h_c L_r \text{ or } S$$

## Component

## Slab

$$\text{Live Load} = 20 \text{ psf} > S = 10 \text{ psf}$$

$$\text{Dead Load} = 14 \text{ psf}$$



2-3

$$w_u = 1.2(166) + 1.6(20) = \boxed{231.2 \text{ psf}}$$

$$w_u = 1.2(166) + 0.5(20) = 209.2 \text{ psf}$$

Column

Live Load - 20 psf Unreducible

Dead Load - 178 psf

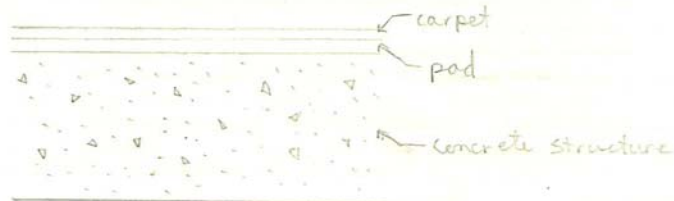
$$w_u = 1.2(178) + 1.6(20) = \boxed{245.6 \text{ psf}}$$

$$w_u = 1.2(178) + 0.5(20) = 223.6 \text{ psf}$$

Answer

2-4

## 1. b. Floor



Cross Section of Typical Floor Construction

## Loads

## Live Loads

Office - 50 psf + 20 psf = 70 psf (IBC 2003)  
 Corridor - 80 psf (IBC 2003) → used to design for building flexibility

## Dead Loads

## Materials

Carpet with Pad (Boise Cascade - Weights of Materials) - 2 psf  
 Superimposed Dead Load - 15 psf

## Structure

slab (10" thk)	$150 \text{ pcf} \cdot \frac{10 \text{ in}}{12 \text{ in}}$	- 12.5 psf
drop panels (4" thk)	$150 \text{ pcf} \cdot \frac{4 \text{ in}}{12 \text{ in}} \cdot \frac{10.67 \cdot 7.33 \text{ ft}^2}{24 \text{ ft} \cdot 32 \text{ ft}}$	- 5 psf
column (30" x 30")	$150 \text{ pcf} \cdot \frac{(168 \text{ in} \cdot 17 \text{ in})}{12 \text{ in}} \cdot \frac{30 \text{ in} \cdot 30 \text{ in}}{12 \text{ in}^2 \cdot 26 \text{ ft} \cdot 32 \text{ ft}}$	- 15 psf

## Load Combinations

1.2D + 1.6L

## Components

## slab

Live Load - 80 psf  
 Dead Load - 142 psf

2-5

$$w_u = 1.2(142) + 1.6(80) = 298.4 \text{ psf}$$

column

$$\text{Live Load} = 80 \text{ psf} < 100$$

$$L = 80 \cdot 0.4$$

$$\text{max } 0.25 + \frac{15}{\sqrt{4 \cdot 26 \cdot 32}} = 0.51 \cdot 80 = 40.8 \text{ psf}$$

$$\hookrightarrow 3328 > 400 \text{ OK } \checkmark$$

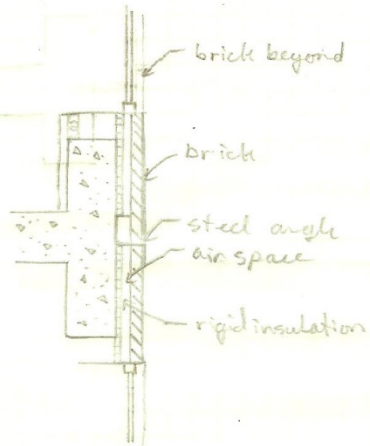
$$\text{Dead Load} = 162 \text{ psf}$$

$$w_u = 1.2 \cdot 162 + 1.6 \cdot 40.8 = 259.7 \text{ psf}$$

ANSWER

2-6

Exterior Wall

Brick Wall

Dead Load

Materials

brick - 42 psf  
 rigid insulation (2" Thk.) - 3 psf

Total Dead Load - 45 psf  $\cdot$  14 ft = 630 plf (for floors 1-3)

- 45 psf  $\cdot$  14.67 ft = 660 plf (for Floor 4)

- 45 psf  $\cdot$  13.115 ft = 590.2 plf (for floor 5)

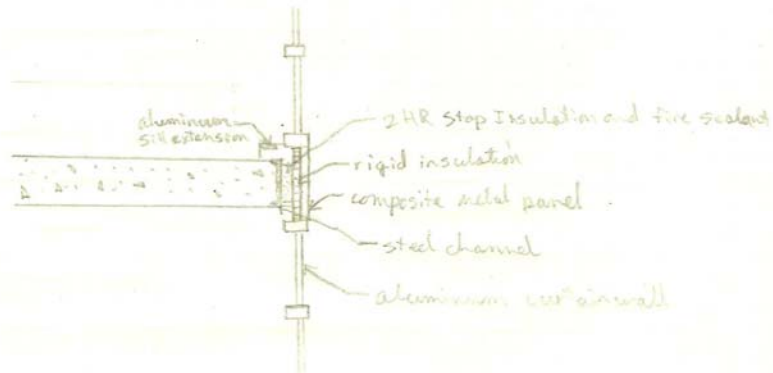
- 45 psf  $\cdot$  12 ft = 540 plf for floor 6 - Roof)

Load transfer

- brick  $\rightarrow$  steel angle  $\rightarrow$  structure

2-7

### Aluminum and Glass Wall



#### Materials

aluminum	-4 psf
rigid insulation (2" thick)	-3 psf
glass (2 pane @ 1/4" thick ea.)	-0 psf

- Total dead load - 15 psf  $\cdot$  14 ft = 210 plf (for floors 1-3)  
 - 15 psf  $\cdot$  14.67 ft = 220 plf (for floor 4)  
 - 15 psf  $\cdot$  13.15 ft = 197 plf (for floor 5)  
 - 15 psf  $\cdot$  12 ft = 180 plf (for floors 6-roof)

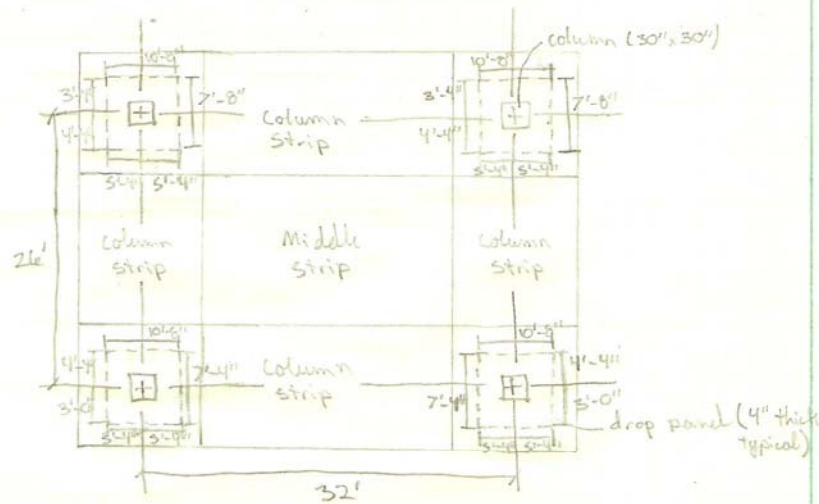
#### Load Transfer

- composite metal panel  $\rightarrow$  steel channel  $\rightarrow$  structure

#### Non-typical Loads

- vehicle impact load - 6 kips at 18" above finish floor
- 6 kips is approximately the weight of a large vehicle
- helicopter pad load - 60 psf on roof between grid 6 and 7 and 8 and 9
- weight includes the weight of the helicopter and the weight of the pad
- ambulance load - 480 plf of load lane - designed as HSI5 from AAS110 - located at the first floor at joists

## Typical Bay



Determine Minimum thickness of slab (with drop panels)

$a_f = 0$  (the interior panel has no perimeter beams)

$$l_n = 32 - \frac{30}{12} = 29.5 \text{ ft}$$

From Table 9.5(c) ACI 318-02

$$\text{Min } h = \frac{l_n}{36} = \frac{29.5}{36} = 0.819 \text{ ft} = 9.83 \text{ in} \approx 10 \text{ in.}$$

Must not be less than 4 in, according to section 9.5.3.2

2-9

Determine Shear Strength of Slab at Column Support

$$f'_c = 5000 \text{ psi}$$

$$f_y = 60,000 \text{ psi}$$

Superimposed Dead Load (Includes self weight of slab) = 298.4 psf

10'-8" x 7'-4" drop panel

Determine the Minimum Size of The Drop Panel based on ACI 318-02 Section 13.2.5.

$$\text{Minimum thickness} = \frac{10 \text{ in}}{4} = 2.5 \times 4 \text{ in}$$

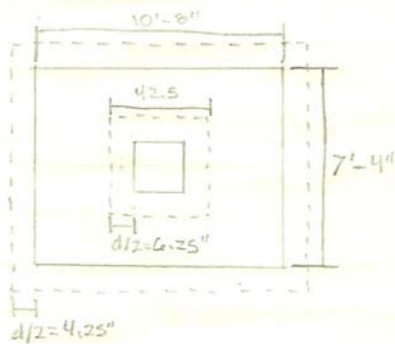
$$\text{Minimum } L/6 = \frac{32}{6} = 5'-4" = 5'-4"$$

$$\text{Minimum } L/6 = \frac{26}{6} = 4'-4" = 4'-4"$$

$$\text{Minimum } L/6 = \frac{20}{6} = 3'-4" = 3'-4"$$

$$\text{Minimum } L/6 = \frac{13.75}{6} = 2'-3\frac{1}{2}" < 3'-0"$$

Critical Section



Investigate shear strength at critical section located  $d/2$  from column perimeter

$$q_{pu} = 1.2 \cdot \frac{4}{12} \cdot 150 = 60 \text{ psf}$$

$$V_u = 0.2984 \text{ ksf} (32 \cdot 26 - 3.54^2) + 0.06 (10.67 \cdot 7.33 - 3.54^2) \\ = 246.47 \text{ kips}$$

2-10

$$b_o = 4 \cdot 42.5 = 170 \text{ in}$$

$$\frac{2 + \frac{4}{B}}{B} = \frac{2 + \frac{4}{1.23}}{1.23} = 5.25 \quad B = \frac{32}{26} = 1.23$$

$$\frac{a_s \cdot d}{b_o} + 2 = \frac{40 \cdot 12.5}{170} + 2 = 4.94$$

min 4

$$\phi V_c = \phi \cdot 4 \cdot \lambda \sqrt{F_c'} b_o d = 0.75 \cdot 4 \cdot 1.0 \cdot \sqrt{5000} \cdot 170 \cdot 12.5 = 450.8 \text{ kips}$$

$$\phi V_c > V_u$$

$$450.8 \text{ kips} > 246.47 \text{ kips OK}$$

Investigate shear strength at critical section  $b_o$  located at  $d/2$  from the edge of the drop panel.

$$V_u = 0.2964 (32 \cdot 26 - 11.375 \cdot 8.04) = 220.98 \text{ kips}$$

$$b_o = 2 \cdot (128 + 8.5) + 2 \cdot (88 + 8.5) = 466 \text{ in}$$

$$\frac{2 + \frac{4}{B}}{B} = \frac{2 + \frac{4}{1.23}}{1.23} = 5.25$$

$$\frac{a_s \cdot d}{b_o} + 2 = \frac{40 \cdot 8.5}{466} + 2 = 2.73$$

min 4

$$\phi V_c = \phi \cdot \left( \frac{a_s \cdot d}{b_o} + 2 \right) \lambda \sqrt{F_c'} b_o d = 0.75 \cdot 2.73 \cdot 1.0 \cdot \sqrt{5000} \cdot 466 \cdot 8.5 = 573.5 \text{ kips}$$

$$\phi V_c > V_u$$

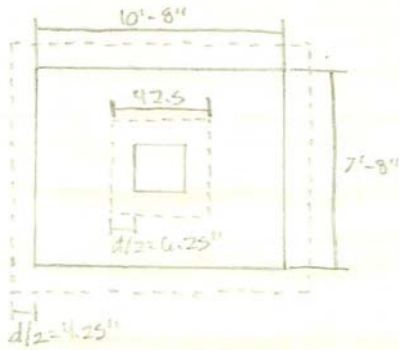
$$573.5 \text{ kips} > 220.98 \text{ kips OK}$$



2-11

10'-8" x 7'-8" drop panel

Critical Section



Investigate shear strength at critical section  $b_o$  located  $d/2$  from column perimeter

$$\phi_{DC} = 1.2 \cdot \frac{4}{12} \cdot 150 = 60 \text{ psf}$$

$$V_u = 0.2984(32.26 - 3.54^2) + 0.06(10.67 \cdot 7.67 - 3.54^2) = 248.7 \text{ kips}$$

$$b_o = 4 \cdot 42.5 = 170 \text{ in}$$

$$2 + \frac{u}{\beta} = 2 + \frac{u}{\beta} = 5.25 \quad \beta = \frac{20}{16} = 1.25$$

$$\frac{1.5 \cdot d}{b_o} + 2 = \frac{1.5 \cdot 12.5}{170} + 2 = 4.94$$

min 4

$$\phi_{VC} = \phi \cdot 4 \cdot \lambda \cdot \sqrt{f'_c} \cdot b_o \cdot d = 0.75 \cdot 4 \cdot 1.0 \cdot \sqrt{5000} \cdot 170 \cdot 12.5 = 450.8 \text{ kips}$$

$$\phi_{VC} > V_u \\ 450.8 \text{ kips} > 248.7 \text{ kips OK}$$

2-12

Investigate shear strength at critical section  $b_o$  located at  $d/2$  from the edge of the drop panel.

$$V_u = 0.2984(32 \cdot 26 - 11.375 \cdot 8 \cdot 375) = 219.8 \text{ kips}$$

$$b_o = 2 \cdot (120 + 8.5) + 2 \cdot (92 + 8.5) = 474$$

$$\frac{2+4}{P} = \frac{2+4}{1.23} = 5.25$$

$$\frac{Q_s \cdot d}{b_o} + 2 = \frac{40 \cdot 8.5}{474} + 2 = 2.72$$

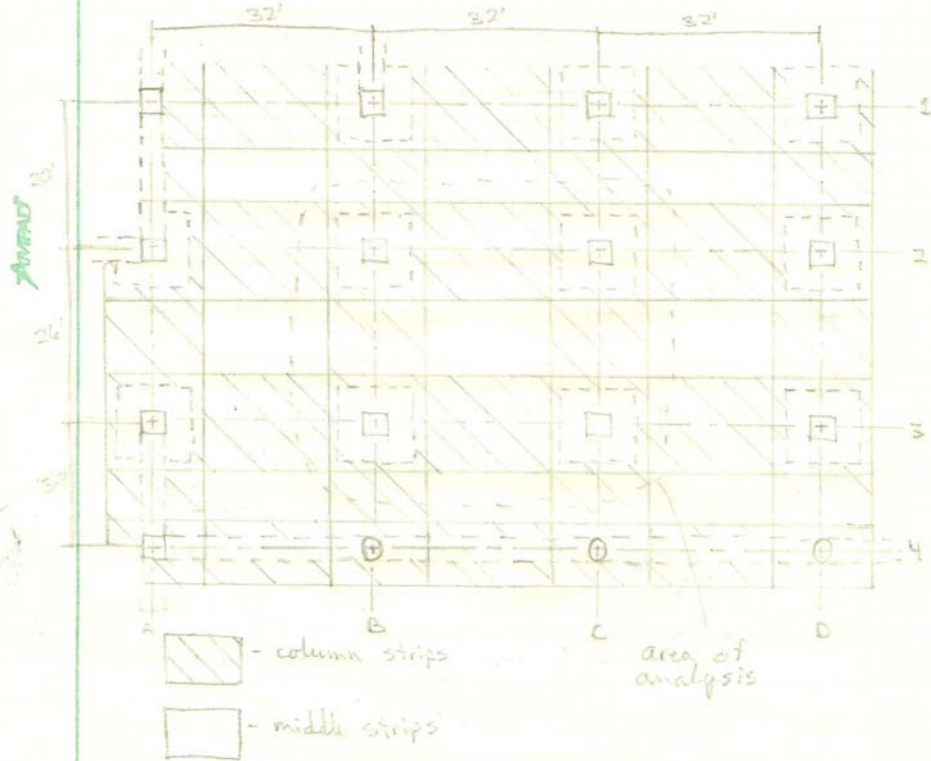
min 4

$$\phi V_c = \phi \left( \frac{Q_s \cdot d}{b_o} + 2 \right) \times \sqrt{f_c'} b_o d = 0.75 \cdot 2.72 \cdot 1.1 \sqrt{3000} \cdot 474 \cdot 8.5 = 581.2 \text{ kips}$$

$$\phi V_c > V_u \\ 581.2 \text{ kips} > 219.8 \text{ kips} \text{ OK}$$

2-13

Determine the positive and negative moments for the column strips and middle strips using the Direct Design Method.



Check if Direct Design Method can be used.

- Minimum of 3 spans in each direction  $\checkmark$
- Ratio of  $l_2 / l_1 \leq 2$

$$\frac{32}{20} = 1.6 \checkmark \quad \frac{32}{26} = 1.23 \checkmark \quad \frac{32}{13.75} = 2.33 \times$$

- Shall not differ by  $\frac{1}{3} l_2$

$$32 - 32 = 0 \checkmark \quad \frac{1}{3} 26 = 8.67 \quad 26 - 8.67 = 17.33 \text{ ft} < 20 \checkmark$$

$$17.33 \text{ ft} > 13.75 \times$$

Note: Based on Section 13.6.1.8, allowed to use Direct Design Method as long as Section 13.5.1 is satisfied.

2-14

Determine moments along gridline B between grids 2 and 3 (Same as moments along gridline C)

Compute  $M_0$ :

$$M_0 = \frac{q_u l_2 l_n^2}{8}$$

$$q_u = 290.4 \text{ psf}$$

$$l_2 = 32 \text{ ft}$$

$$l_n = 26' - \frac{30}{12}$$

$$M_0 = \frac{290.4 \cdot 32 \cdot 26.5^2}{8} = 659.2 \text{ ft}\cdot\text{k}$$

Divide  $M_0$  into negative and positive moments:

$$\text{Negative moment} = -0.65 M_0 = -428.48 \text{ ft}\cdot\text{k}$$

$$\text{Positive moment} = 0.35 M_0 = 230.72 \text{ ft}\cdot\text{k}$$

Divide moments between the column and middle strips

Negative moments

Column-strip negative moment =

$$l_2/l_1 = 32/26 = 1.23$$

$$\alpha_f = 0 \text{ (no beams)}$$

$$0.75 \cdot -428.48 \text{ ft}\cdot\text{k} = -321.36 \text{ ft}\cdot\text{k}$$

Middle-strip negative moment =

$$0.25 \cdot -428.48 = -107.12 \text{ ft}\cdot\text{k}$$

Positive Moments

Column-strip positive moment =

$$l_2/l_1 = 1.23$$

$$\alpha_f = 0 \text{ (no beams)}$$

$$0.6 \cdot 230.72 \text{ ft}\cdot\text{k} = 138.4 \text{ ft}\cdot\text{k}$$

Middle-strip positive moment =

$$0.4 \cdot 230.72 \text{ ft}\cdot\text{k} = 92.29 \text{ ft}\cdot\text{k}$$

2-15

Determine moments along gridline 2 between grids B and C

Compute  $M_o$ :

$$M_o = \frac{q_u l_2 l_n^2}{8}$$

$$q_u = \frac{298.4}{2} \text{ psf}$$

$$l_2 = \frac{20 + 26}{2} = 23 \text{ ft}$$

$$l_n = 32 - \frac{20}{2} = 29.5 \text{ ft}$$

$$M_o = \frac{298.4 \cdot 23 \cdot 29.5^2}{8} = 796.6 \text{ ft}\cdot\text{k}$$

Divide  $M_o$  into negative and positive moments:

$$\text{Negative Moment} = 0.65 M_o = 485.3 \text{ ft}\cdot\text{k}$$

$$\text{Positive Moment} = 0.35 M_o = 261.3 \text{ ft}\cdot\text{k}$$

Divide Moments between column strips and middle strips

Negative moments

Column-strip negative moment:

$$l_2/l_1 = 23/32 = 0.72$$

$\alpha_1 = 0$  (no beams)

$$0.75 \cdot 485.3 \text{ ft}\cdot\text{k} = -363.98 \text{ ft}\cdot\text{k}$$

Middle-strip negative moment:

$$0.25 \cdot 485.3 \text{ ft}\cdot\text{k} = -121.33 \text{ ft}\cdot\text{k}$$

Positive Moments

Column-strip positive moment:

$$l_2/l_1 = 0.72$$

$\alpha_1 = 0$  (no beams)

$$0.6 \cdot 261.3 \text{ ft}\cdot\text{k} = 156.78 \text{ ft}\cdot\text{k}$$

Middle-strip positive moment:

$$0.4 \cdot 261.3 \text{ ft}\cdot\text{k} = 104.52 \text{ ft}\cdot\text{k}$$

2-16

Determine moments along gridline 3 between grids B and C

Compute  $M_0 =$

$$M_0 = \frac{q_u \cdot l_2 \cdot l_n^2}{8}$$

$$q_u = 298.4 \text{ psf}$$

$$l_2 = \frac{26.11 \text{ ft}}{1.32} = 19.86 \text{ ft}$$

$$l_n = 32 - \frac{11}{12} = 29.5 \text{ ft}$$

$$M_0 = \frac{298.4 \cdot 19.86 \cdot 29.5^2}{8} = 645.31 \text{ ft}\cdot\text{k}$$

Divide  $M_0$  into negative and positive moments:

$$\text{Negative Moment} = -0.65 M_0 = -419.45 \text{ ft}\cdot\text{k}$$

$$\text{Positive Moment} = 0.35 M_0 = 225.86 \text{ ft}\cdot\text{k}$$

Divide Moments between column strips and middle strips

Negative moments

Column-strip negative moment-

$$l_2/l_n = 19.86/32 = 0.62$$

$a_f = 0$  (no beams)

$$0.75 \cdot -419.45 \text{ ft}\cdot\text{k} = -314.59 \text{ ft}\cdot\text{k}$$

Middle-strip negative moment =

$$0.25 \cdot -419.45 \text{ ft}\cdot\text{k} = -104.86 \text{ ft}\cdot\text{k}$$

Positive Moments

Column-strip positive moment

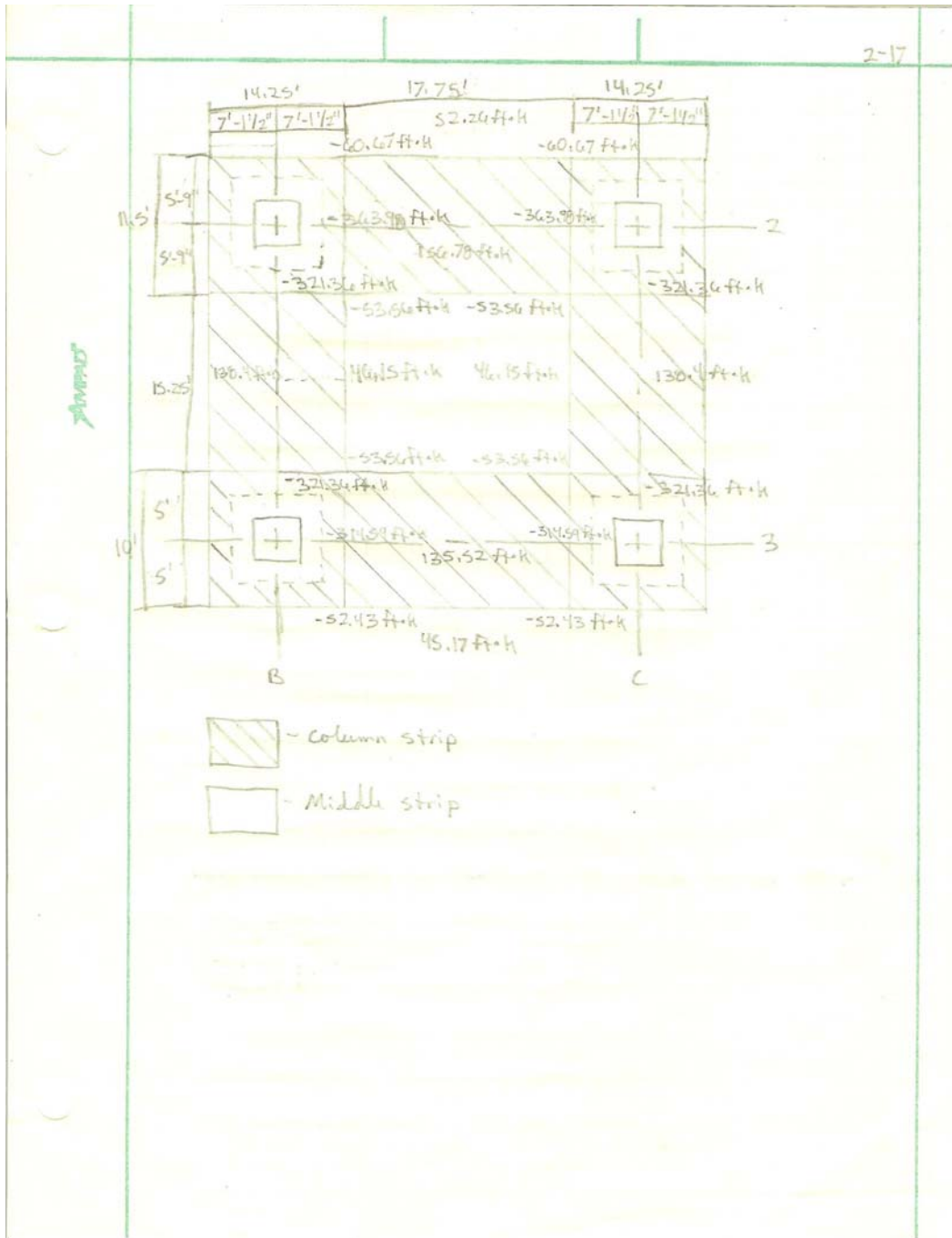
$$l_2/l_n = 0.62$$

$a_f = 0$  (no beams)

$$0.6 \cdot 225.86 \text{ ft}\cdot\text{k} = 135.52 \text{ ft}\cdot\text{k}$$

Middle-strip positive moment

$$0.4 \cdot 225.86 \text{ ft}\cdot\text{k} = 90.34 \text{ ft}\cdot\text{k}$$



2-16

Compute the area of steel along grid 2

$$A_s = \frac{M_u}{\phi F_y j d} \quad (\text{Eq. 13-44 from Wright and MacGregor})$$

$$d = h - 0.75 - 0.75/2 = 8.875 \text{ in}$$

$$\text{assume } j = 0.95$$

$$A_{s(\text{req'd})} = \frac{363.98 \text{ ft} \cdot \text{k} \cdot 12,000}{0.9 \cdot 60,000 \cdot 0.95 \cdot 8.875} = 9.59 \text{ in}^2$$

$$a = \frac{A_s f_y}{\phi F_c b} = \frac{9.59 \cdot 60,000}{0.85 \cdot 5000 \cdot 11.5 \cdot 12} = 0.98 \text{ in}$$

$$c = \frac{a}{\beta_1} = \frac{0.98}{0.80} = 1.23 \text{ in}$$

$$g = 0.98$$

$$\beta_1 = 0.85 - \frac{g - 0.25}{1000} (f_c' - 4000) \geq 0.65$$

$$0.85 - \frac{0.98 - 0.25}{1000} (5000 - 4000) = 0.8 \geq 0.65$$

$$1.23 \leq \frac{3g}{8} = 3.33 \text{ in} \therefore \phi = 0.9$$

$$A_s = \frac{M_u \cdot 12,000}{0.9 \cdot 60,000 \cdot (8.875 - \frac{0.98}{2})} \quad (\text{assuming } a \text{ is constant for all sections})$$

$$A_s = 0.0265 M_u$$

$$A_{s, \text{min}} = 0.0018 b h$$

$$b = 11.5 \text{ ft}$$

$$h = 10 \text{ in}$$

$$A_{s, \text{min}} = 11.5 \cdot 12 \cdot 0.0018 \cdot 10 = 2.48 \text{ in}^2$$

Compute the area of steel along grid 3

$$A_s = \frac{M_u}{\phi F_y j d}$$

$$d = h - 0.75 - 0.75/2 = 8.875 \text{ in}$$

$$\text{assume } j = 0.95$$

$$A_{s(\text{req'd})} = \frac{314.59 \text{ ft} \cdot \text{k} \cdot 12,000}{0.9 \cdot 60,000 \cdot 0.95 \cdot 8.875} = 8.21 \text{ in}^2$$



2-19

$$a = \frac{A_s f_y}{0.85 f_c' b} = \frac{8.27 \cdot 60,000}{0.85 \cdot 5000 \cdot 10 \cdot 12} = 0.98 \text{ in}$$

$$c = \frac{a}{\beta_1} = \frac{0.98}{0.80} = 1.23 \text{ in}$$

$$a = 0.98$$

$$\beta_1 = 0.80$$

$$1.23 < \frac{3d}{8} = 3.33 \therefore \phi = 0.9$$

$$A_s = \frac{M_u \cdot 12,000}{0.9 \cdot 60,000 \cdot \left(8.675 - \frac{0.98}{2}\right)} \quad (\text{assuming } a \text{ is constant for all sections})$$

$$A_s = 0.0265 M_u$$

$$A_{s, \min} = 0.0018 b h$$

$$b = 10 \text{ ft}$$

$$h = 10 \text{ in}$$

$$A_{s, \min} = 0.0018 \cdot 10 \cdot 10 \cdot 12 = 2.16 \text{ in}^2$$

Computer the area of steel along grid B and C

$$A_s = \frac{M_u}{\phi f_y j d}$$

$$d = h - 0.75 - 0.75 - 0.75/2 = 8.125 \text{ in}$$

$$\text{assum } j = 0.95$$

$$A_s (\text{req'd}) = \frac{321.36 \text{ ft} \cdot 11 \cdot 12,000}{0.9 \cdot 60,000 \cdot 0.95 \cdot 8.125} = 9.25 \text{ in}^2$$

$$a = \frac{A_s f_y}{0.85 f_c' b} = \frac{9.25 \cdot 60,000}{0.85 \cdot 5000 \cdot 14.25 \cdot 12} = 0.76 \text{ in}$$

$$c = \frac{a}{\beta_1} = \frac{0.76}{0.80} = 0.95$$

$$\beta_1 = 0.80$$

$$a = 0.76$$

$$\beta_1 = 0.80$$

$$0.95 < \frac{3d}{8} = 3.05 \therefore \phi = 0.9$$

$$A_s = \frac{M_u \cdot 12,000}{0.9 \cdot 60,000 \cdot \left(8.125 - \frac{0.76}{2}\right)}$$

$$A_s = 0.0287 M_u$$

$$A_{s, \min} = 0.0018 b h$$

$$b = 14.25$$

$$h = 10 \text{ in}$$

$$A_{s, \min} = 0.0018 \cdot 14.25 \cdot 12 \cdot 10 = 3.09 \text{ in}^2$$

2-20

Grid B and C	Column Strip	Middle Strip
Interior Negative Moment		
Total strip moment (kip-ft)	-321.36	-107.12
Required $A_s$ (in <sup>2</sup> )	9.22	3.07
Minimum $A_s$ (in <sup>2</sup> )	3.08	3.83
Selected steel	#6@6inoc	#6@12inoc
As provided (in <sup>2</sup> )	12.32	7.48

Grid B and C	Column Strip	Middle Strip
Interior Positive Moment		
Total strip moment (kip-ft)	139.4	92.3
Required $A_s$ (in <sup>2</sup> )	3.97	2.65
Minimum $A_s$ (in <sup>2</sup> )	3.08	3.83
Selected steel	#6@7inoc	#6@12inoc
As provided (in <sup>2</sup> )	10.56	7.48

Grid 1	Column Strip	Middle Strip
Interior Negative Moment		
Total strip moment (kip-ft)	-363.98	-121.33
Required $A_s$ (in <sup>2</sup> )	9.65	3.22
Minimum $A_s$ (in <sup>2</sup> )	2.46	3.29
Selected steel	#6@6inoc	#6@12inoc
As provided (in <sup>2</sup> )	10.12	6.6

Grid 2	Column Strip	Middle Strip
Interior Positive Moment		
Total strip moment (kip-ft)	156.76	104.52
Required $A_s$ (in <sup>2</sup> )	4.15	2.77
Minimum $A_s$ (in <sup>2</sup> )	2.48	3.29
Selected steel	#6@7inoc	#6@12inoc
As provided (in <sup>2</sup> )	10.12	6.6

Grid 3	Column Strip	Middle Strip
Interior Negative Moment		
Total strip moment (kip-ft)	-314.51	-104.84
Required $A_s$ (in <sup>2</sup> )	8.34	2.78
Minimum $A_s$ (in <sup>2</sup> )	2.16	3.29
Selected steel	#6@6inoc	#6@12inoc
As provided (in <sup>2</sup> )	8.8	6.6

summary

2-21

Grid 3 Interior Positive Moment	Column Strip	Middle Strip
Total Strip Moment (kip-ft)	135.52	90.34
Required $A_s$ (in <sup>2</sup> )	3.59	2.39
Minimum $A_s$ (in <sup>2</sup> )	2.16	3.29
Selected steel As provided (in <sup>2</sup> )	#6@7in o.c. 6.6	#6@6in o.c. 6.6



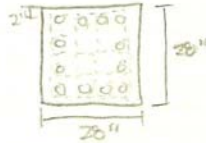
e-22

## Interior Column

$$f'_c = 6000 \text{ psi}$$

$$f_y = 60,000 \text{ psi}$$

$$A_{st} = 9.48$$



$$P_u = [1.2(178) + 0.5(20) + 1.2(162) \cdot 12 + 1.6(40.8) \cdot 12] \cdot 736$$

$$= 2458 \text{ k}$$

$$P_o = 0.85 f'_c (A_g - A_{st}) + f_y A_{st}$$

$$= 0.85 \cdot 6000 \cdot (784 - 9.48) + 60,000 \cdot 9.48$$

$$= 4518$$

$$\phi P_o = 4518 \cdot 0.65 = 2936.7 \text{ k} > 2458 \text{ k} \quad \checkmark$$

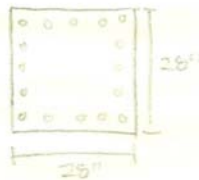
## Exterior Column

$$f'_c = 6000 \text{ psi}$$

$$f_y = 60,000 \text{ psi}$$

$$A_{st} = 20.32$$

$$d' = 3.01 \text{ in.}$$



$$L = 80 \left( 0.25 + \frac{15}{\sqrt{32 \cdot 2 \cdot 29.29}} \right) = 80 \cdot 0.596 = 47.68 \text{ psf}$$

includes edge beam

$$P_u = [1.2(178) + 0.5(20) + 1.2(206) \cdot 12 + 1.6(47.68) \cdot 12] \cdot 968.64$$

$$+ 1.2(6920.2)$$

$$= 1932.3 \text{ k}$$

$$M_u = \frac{w_u L^2}{12} = \frac{298.4 \cdot 29.29^2}{12} = 728.3 \text{ k} \cdot \text{ft}$$

$$\gamma = \frac{h - 2d'}{h} = \frac{28 - 2(3.01)}{28} = 0.785$$

$$p_g = \frac{6 \cdot 1.27}{28 \cdot 28} = 0.026 = 2.6\%$$

Using Figure A-11a and A-11b From Wright and MacGregor

$$\text{for } \gamma = 0.75 \quad \phi P_n = 3.35 b \cdot h$$

$$\text{for } \gamma = 0.9 \quad \phi P_n = 3.35 b \cdot h$$

$$\phi P_o = 3.35 \cdot \frac{b \cdot h}{0.8} = 3.35 \cdot \frac{28 \cdot 28}{0.8} = 3283 \text{ k} > 1932.3 \text{ k} \quad \checkmark$$

2-23

$$\text{for } \gamma = 0.75 \quad \phi M_n = 0.54 \cdot b \cdot h^2$$

$$\text{for } \gamma = 0.90 \quad \phi M_n = 0.42 \cdot b \cdot h^2$$

$$\phi M_n = 0.58 \cdot 28 \cdot 28^2 = 1061 \text{ k}\cdot\text{ft} + 7728.3 \text{ k}\cdot\text{ft} \checkmark$$

\*over

2-24

Alternate System 1: Non-composite steel

Loading

Live Load

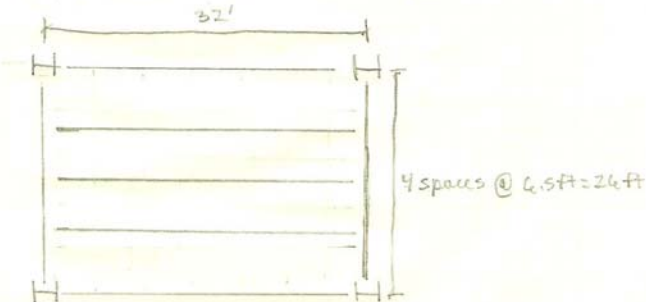
office =  $50 + 20 = 70$  psf

corridor = 80 psf → used for building flexibility

Dead load

carpet with pad = 2 psf

Superimposed Dead Load = 15 psf



Determine Deck Size

Use composite decking - more commonly used with rolled beams and girders.

Use light weight concrete for lighter construction

Use a topping of  $3/4$ " for 2 hour fire rating

Use unshored construction for more economical design

Try 1.5 VLR 22 gauge

3 span unshored construction

- 7'7" → 6'6"

Superimposed Load = 97 psf

Superimposed Live Load for 6'-6" span = 260 psf

 $260 > 97$ Use Valcraft 1.5 VLR 22 gauge deck with  $3/4$ " LW topping

2-25

## Determine Beam Size

## Load

Live Load - 80 psf (reducible)

$$L = L_o \left( 0.25 + \frac{15}{\sqrt{\text{ft}} \cdot \text{ft}} \right) \geq 0.5 L_o$$

$$= 80 \cdot \left( 0.25 + \frac{15}{\sqrt{13 \cdot 22}} \right) = 80 \cdot 0.986 = 78.8 \text{ psf}$$

↳ 4% > 400 ✓

## Dead Load

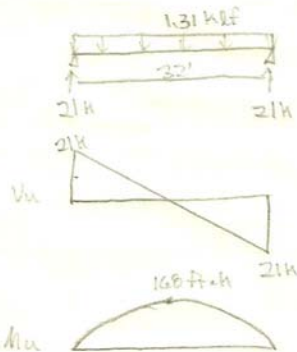
Deck - 41 psf  
 carpet with pad - 2 psf  
 Superimposed dead load - 15 psf  
 beam self weight allowance - 3 psf

Total - 63 psf

$$w_u = 1.2(63) + 1.6(78.8) = 201.68 \text{ psf}$$

$$w_u = \frac{201.68 \cdot 6.5}{1000} = 1.31 \text{ klf}$$

## Shear and Moment Diagrams



2-26

## Check Bending

Using Zx tables in AISC 14th ed.  
 $- M_u \leq \phi_b M_{px}$

$$W14 \times 30 - \phi_b M_{px} = 177 \text{ Ft}\cdot\text{h} > 165 \text{ Ft}\cdot\text{h}$$

$$W16 \times 31 - \phi_b M_{px} = 203 \text{ Ft}\cdot\text{h} > 165 \text{ Ft}\cdot\text{h}$$

## Check Shear

Using Zx tables:  
 $- V_u \leq \phi_v V_{nx}$

$$W14 \times 30 - \phi_v V_{nx} = 112 \text{ kips} > 21 \text{ kips}$$

$$W16 \times 31 - \phi_v V_{nx} = 131 \text{ kips} > 21 \text{ kips}$$

## Check Deflection

$- \Delta_{LL} \leq L/360$

W14x30

$$\Delta_{LL} = \frac{5 w_{LL} L^4}{384 E I_x} = \frac{5 (0.517) (32')^4 (1728)}{384 \cdot 29000 \cdot 291} = 1.43 \text{ in}$$

$$L/360 = \frac{32 \cdot 12}{360} = 1.067 < 1.43 \text{ in No Good}$$

$$I_{requ} = 291 \cdot \frac{1.43}{1.067} = 390 \text{ in}^4$$

Using Ix tables

$$W18 \times 35 - I_x = 510 \text{ in}^4 > 390 \text{ in}^4$$

## Check Beam weight assumption

$$\text{weight of beam} - \frac{35}{2.5} = 5.38 \text{ psf} > 5 \text{ psf assumed}$$

## Recheck Bending with new loads

Loading

Live Load

$$- 78.8 \text{ psf (reduced)}$$



2-27

## Dead Load

Deck - 41 psf  
 carpet with pad - 2 psf  
 Superimposed dead Load - 15 psf  
 beam self weight allowance - 6 psf

$$\text{Total} = 64 \text{ psf}$$

$$w_u = 1.2(64) + 1.6(79.8) = 202.88 \text{ psf}$$

$$w_u = \frac{202.88 \cdot 6.5}{1000} = 1.32 \text{ klf}$$

## shear and Moment

$$V_u = 21.12 \text{ k}$$

$$M_u = 169 \text{ ft} \cdot \text{k}$$

## check Bending

using 3x tables

$$W18 \times 35 - \phi_b M_{px} = 249 \text{ k} \cdot \text{ft} > 169 \text{ k} \cdot \text{ft} \checkmark$$

## check Shear

Using 3x tables

$$W18 \times 35 - \phi_v V_{nx} = 159 \text{ k} > 21.12 \text{ k} \checkmark$$

Use W18x35 beams

Answer

2-28

## Determine Girder Size

## Load

Live Load - 80 psf (reducible)

$$L = L_0 \left( 0.25 + \frac{15}{\sqrt{FLLA}} \right) \geq 0.5L$$

$$= 80 \left( 0.25 + \frac{15}{\sqrt{26 \cdot 32 \cdot 2}} \right) = 80 \cdot 0.618 = 49.4 \text{ psf}$$

## Dead Load

Deck - 41 psf

carpet with pad - 2 psf

superimposed dead load - 15 psf

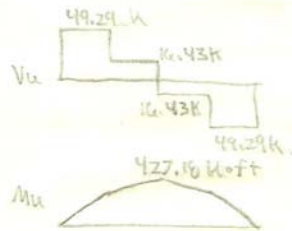
beam self weight allowance - 6 psf

Girder self weight allowance - 2 psf

$$W_u = 1.2(66 \text{ psf}) + 1.6(49.4 \text{ psf}) = 158 \text{ psf}$$

$$P_u = 158 (6.5) (3) = 32.86 \text{ k}$$

## Shear and Moment Diagrams



## Check Bending

Using  $Z_x$  tables  
- Max  $\leq \phi M_p$ 

$$W24 \times 55 - \phi M_p = 473 \text{ k-ft} \geq 427.16 \text{ k-ft}$$

2-29

## Check Shear

Using  $Z_x$  tables  
 $-V_u \leq \phi_v V_{ux}$

$$W 18 \times 55 - \phi_v V_{ux} = 234 \text{ k} > 49.29 \text{ k}$$

## Check Deflections

$$\Delta_{LL} \leq L/360$$

$$\Delta_{LL} = \frac{P_{LL} L^3}{288 EI_x} = \frac{10.28 \cdot 26^3 \cdot 1728}{288 \cdot 29000 \cdot 1140} = 0.337 \text{ in}$$

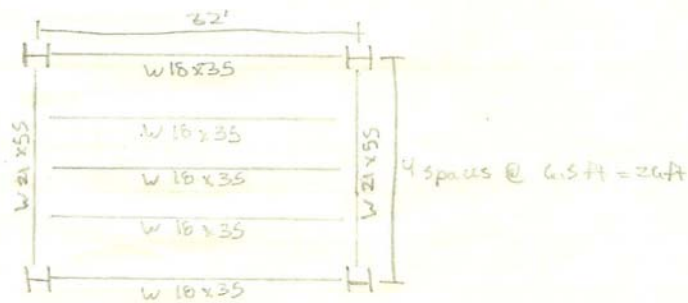
$$L/360 = \frac{26 \cdot 12}{360} = 0.867 \text{ in} > 0.337 \text{ in}$$

## Check girder weight assumption

$$\text{weight of girder} = 5.35 \text{ psf for beams} + \frac{55}{82} = 7.1 \text{ psf} < 6 \text{ psf}$$

OK

Use  $w 21 \times 55$  for girders



2-30

alternate system 2: Composite Steel

Loading

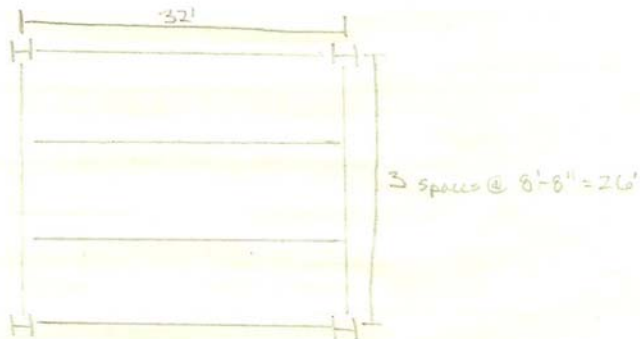
Live Load

office =  $50 + 20 = 70$  psfcorridor = 80 psf  $\rightarrow$  used for building flexibility

Dead Load

carpet with pad = 2 psf

Superimposed dead load = 15 psf



Determine Deck size

Use composite decking - more commonly used with rolled beams and girders

Use normal weight concrete

Use a topping of 4 1/2" for 2 hour fire rating

Use unshored construction for more economical design

Try 2 VLI 22 gauge

3 span unshored construction

- 7'-11" < 8'-8"  $\rightarrow$  use higher gauge

Try 2 VLI 20 gauge

3 span unshored construction

- 9'-0" &gt; 8'-8"

2-31

Superimposed Load - 97 psf

Superimposed Live Load for 9'-0" span - 213 psf

213 psf &gt; 97 psf

Use Vulcraft 2VLI 20 gauge deck with 4 1/2" WW topping

Determine Beam Size

Load

Live Load - 80 psf (reducible)

$$L = L_o \left( 0.25 + \frac{15}{\sqrt{KLLAT}} \right) \geq 0.5 L_o$$

$$= 80 \left( 0.25 + \frac{15}{\sqrt{17.333 \cdot 32}} \right) = 80 \cdot 0.887 = 70.96 \text{ psf}$$

↘ 555 > 400 ✓

Dead Load

Deck - 69 psf

carpet with pad - 2 psf

Superimposed dead load - 15 psf

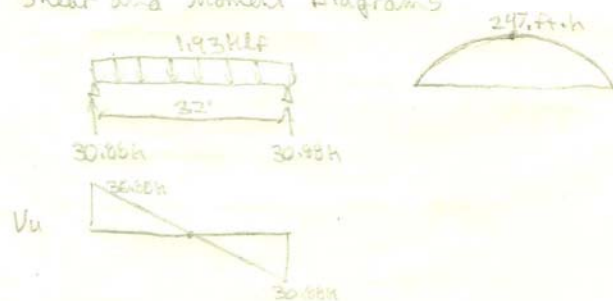
beam self weight allowance - 5 psf

total - 91 psf

$$w_u = 1.2(91) + 1.6(70.96) = 222.7 \text{ psf}$$

$$w_u = \frac{222.7 \cdot 8.67}{1000} = 1.93 \text{ klf}$$

Shear and Moment Diagrams



2-32

- Determine  $\phi M_n$  and  $\phi Q_n$   
 Use Table 3-19 from AISC 14th ed.  
 - assume  $f'_c = 4000$  psi  
 - assume  $a = 2$   
 - assume deck is perpendicular  
 - assume  $\frac{1}{2}$  inch studs per stud  
 - assume 3/4 stud  
 - assume 1 stud per foot

$$I + a = 2; y = 6.50 - \frac{1}{2} = 6''$$

$$W 14 \times 26 \quad \phi Q_n = 135 = \frac{135}{17.2} = 7.85 \Rightarrow 8 \times 2 = 16 \text{ studs/beam}$$

$$\phi M_n = 255 \text{ ft}\cdot\text{k}$$

Next  $\rightarrow$   $W 16 \times 26 \quad \phi Q_n = 96 = \frac{96}{17.2} = 5.58 \Rightarrow 6 \times 2 = 12 \text{ studs/beam}$   
 economical  $\phi M_n = 252$

by inspection  $W 12 \times 26 \quad \phi Q_n = 198 = \frac{198}{17.2} = 11.5 \Rightarrow 12 \times 2 = 24 \text{ studs/beam}$   
 $\phi M_n = 269$

$W 12 \times 22 \quad \phi Q_n = 238 = \frac{238}{17.2} = 13.8 \Rightarrow 14 \times 2 = 28 \text{ studs/beam}$   
 $\phi M_n = 263$

$$a = \frac{\phi Q_n}{0.85 f'_c b_e f}$$

$$b_e f = \frac{32 \cdot 12}{8} = 48 \quad \left| \quad \frac{32 \cdot 12}{8} = 48 \right.$$

$$\min \left| \frac{8.67 \cdot 12}{2} = 52.02 \quad \min \left| \frac{8.67 \cdot 12}{2} = 52.02 \right.$$

$$b_e f = 48 + 48 = 96 \text{ in}$$

$$a = \frac{96}{0.85 \cdot 4 \cdot 96} = 0.299 \therefore \text{if actual} = 6.50 - \frac{0.299 \cdot 12}{2} = 6.35 \Rightarrow 6.5''$$

$\therefore$  assumption incorrect

Use  $y = 6.5''$   
 $\phi Q_n = 96 \Rightarrow 12 \text{ studs/beam}$   
 $\phi M_n = 255 \text{ ft}\cdot\text{k}$

Check unshored strength

$$W 16 \times 26, \phi M_n = 262 \text{ ft}\cdot\text{k}$$

$$w_u = 1.4(69)(8.67) + 1.4(26) = 0.874 \text{ k/ft}$$

$$w_u = 1.2 \left[ (69)(8.67) + 26 \right] + 1.6(20)(8.67) = 1.03 \text{ k/ft} \rightarrow \text{controls}$$

$$M_u = \frac{0.87 \cdot 32^2}{8} = 131.6 \text{ ft}\cdot\text{k} < 262 \text{ ft}\cdot\text{k} \rightarrow \text{ok for unshored construction}$$

2-33

Check wet concrete deflection

$$w_{wc} = 69(0.67) + 26 = 0.624 \text{ klf}$$

$$\Delta_{wc} = \frac{5(0.624)(32)^4(1728)}{384(29000)(199)} = 1.69 \text{ in}$$

$$\Delta_{wc \text{ max}} = \frac{32 \cdot 12}{240} = 1.6 \text{ in} + 1.69 \text{ in} \text{ need to chamber by } 1.5''$$

$$0.8(1.69) = 1.5''$$

Check Live Load Deflection

$$w_{LL} = 70.96 \cdot 0.67 = 0.615 \text{ klf}$$

$$I_{LB} @ y = 615 \text{ and } 20n = 96$$

$$I_{LB} = 617$$

$$\Delta_{LL} = \frac{5(0.615)(32)^4(1728)}{384(29000)(617)} = 0.811 \text{ in}$$

$$\Delta_{LL \text{ max}} = \frac{L}{340} = \frac{32 \cdot 12}{340} = 1.07 \text{ in} > 0.811 \text{ in} \text{ ok}$$

Use W16x26 with 12 studs per beam for beams

Determine Girder Size

Load

Live Load = 80 psf (reducible)

$$L = L_o \left( 0.25 + \frac{15}{\sqrt{f_{LL}}} \right) \geq 0.5 L_o$$

$$= 80 \left( 0.25 + \frac{15}{\sqrt{\frac{24 \cdot 32 \cdot 2}{1664 \cdot 7400}}} \right) = 80 \cdot 0.618 = 49.44 \text{ psf}$$

Dead Load

Deck = 49 psf

carpet with pad = 7 psf

Super imposed dead load = 15 psf

beam self weight allowance = 5 psf

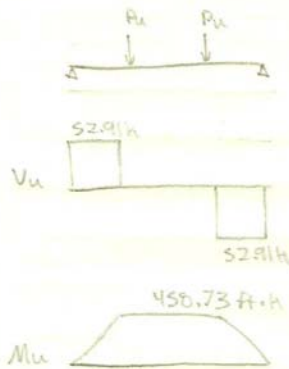
Girder self weight allowance = 2 psf

2-34

$$w_u = 1.2(93) + 1.6(49.44) = 190.7 \text{ psf}$$

$$P_u = 190.7(0.67)(32) = 52.91 \text{ k}$$

Shear and Moment Diagrams

Determine  $\phi M_n$  and  $\phi Q_n$   
Use table 3-19

- assume  $F_c = 4000 \text{ psi}$
- assume  $y = 6.5''$
- assume deck is parallel
- assume 1 weak stud per r/c
- assume  $3/4''$  stud
- assume 1 stud per foot

$$y = 6.5''$$

$$W18 \times 46 \quad \phi Q_n = 169 = \frac{169}{21.5} = 7.84 \Rightarrow 8 \times 2 = 16 \text{ studs per girder}$$

$$\phi M_n = 507 \text{ k-ft} \quad \frac{21.5}{12} = 1.8 \text{ ft} \Rightarrow w/h_r = 5/2 = 2.5$$

$$W18 \times 40 \quad \phi Q_n = 211 = \frac{211}{21.5} = 9.81 \Rightarrow 10 \times 2 = 20 \text{ studs per girder}$$

$$\phi M_n = 486 \text{ k-ft} \quad \frac{21.5}{12} = 1.8 \text{ ft}$$

$$W16 \times 40 \quad \phi Q_n = 237 = \frac{237}{21.5} = 11.02 \Rightarrow 12 \times 2 = 24 \text{ studs per girder}$$

$$\phi M_n = 463 \text{ k-ft} \quad \frac{21.5}{12} = 1.8 \text{ ft}$$

Most economical  $W18 \times 35 \quad \phi Q_n = 260 = \frac{260}{21.5} = 12.09 \Rightarrow 13 \times 2 = 26 \text{ studs per girder}$   
by inspection  $\phi M_n = 465 \text{ k-ft} \quad \frac{21.5}{12} = 1.8 \text{ ft}$



2-35

check unshored strength

$$P_u = [1.2[(69 \cdot 8.67) + 35] + 1.6(20 \cdot 8.67)] 32 = 31.12k$$

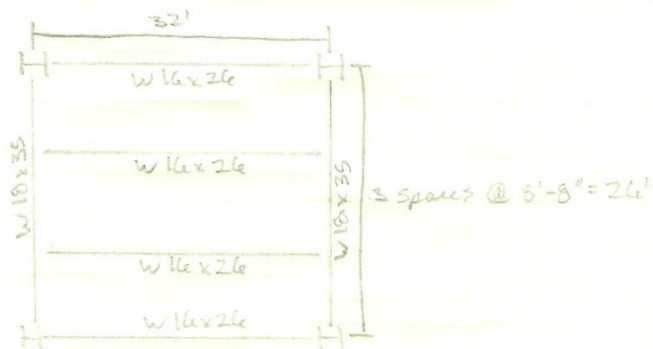
$$M_u = 31.12k \cdot 8.67 = 269.81 \text{ ft} \cdot k < 465k \cdot \text{ft}$$

check Live Load Deflection:

$$\Delta_{LL} = \frac{13.72 \cdot 26^3 \cdot 1728}{28 \cdot 29000 \cdot 1320} = 0.389 \text{ in}$$

$$\Delta_{allow} = \frac{l}{360} = \frac{26 \cdot 12}{360} = 0.867 \text{ in} > 0.389 \text{ in OK}$$

Use W18x35 with 24 studs per girder for girders



2-36

alternate system 3: one way slab with beams and girders

Loading

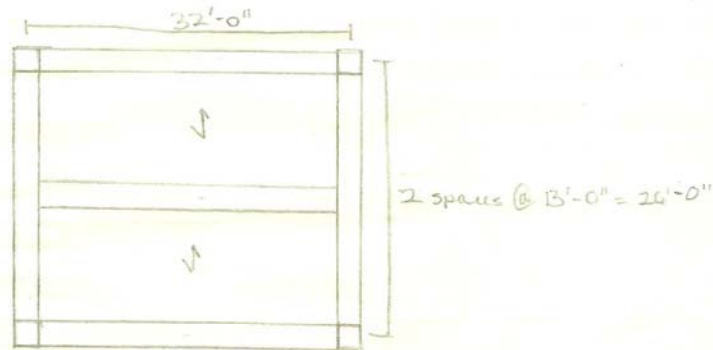
Live Load

office =  $50 + 20 = 70$  psfcorridor = 80 psf  $\rightarrow$  used for building flexibility

Dead Load

carpet with pad = 2 psf

superimposed dead load = 15 psf



Determine slab size

assume  $F'_c = 5000$  psiassume  $F_y = 60$  ksi

Minimum Thickness of Slab

Using ACI 318-11 Table 9.5(a)

$$h_{min} = \frac{l}{20} = \frac{13 \cdot 12}{20} = 5.57 \text{ in} \Rightarrow 6 \text{ in}, d = 5 \text{ in}$$

Loads

Live Loads = 80 psf

Dead Loads

carpet with pad = 2 psf

superimposed dead load = 15 psf

slab dead load =  $\frac{1}{12} \cdot 150 = 7.5$  psf

total = 92 psf

$$w_u = 1.2(92) + 1.6(80) = 238.4 \text{ psf}$$

2-37

Determine design moment

$$M_u = \frac{w_u l^2}{12} = \frac{238.4 \cdot 13^2}{12} = 3357.5 \text{ lb}\cdot\text{ft} = 3.36 \text{ k}\cdot\text{ft}$$

Estimate  $A_s$ 

$$A_s = M_u / f_y d = \frac{3.36}{4.5} = 0.166 \text{ in}^2/\text{ft} \Rightarrow \text{Provide } \approx 4 \text{ bar @ } 12'' \text{ o.c.} \\ (0.2 \text{ in}^2/\text{ft})$$

$$d = 6'' - 0.75'' - \frac{0.375''}{2} = 5''$$

Check  $\phi M_n > M_u$ Assume  $\epsilon_s > \epsilon_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 \cdot f_c' \cdot b} = \frac{0.2 \cdot 60}{0.85 (5) (12)} = 0.235''$$

$$c = \frac{a}{\beta_1} = \frac{0.235}{0.8} = 0.294''$$

$$\beta_1 = 0.85 - \frac{f_c' - 4000}{1000} (5000 - 4000) = 0.8 \geq 0.65$$

$$\epsilon_s = \frac{f_y}{E_s} (d - c) = \frac{60,000}{29,000,000} (5 - 0.294) = 0.0107 > 0.005 \Rightarrow \phi = 0.9$$

$$\phi M_n = \phi A_s \cdot f_y (d - a/2) = 0.9 (0.2) (60) (5 - \frac{0.235}{2}) = 4.39 \text{ k}\cdot\text{ft} > 3.36 \text{ k}\cdot\text{ft}$$

Temperature and shrinkage reinforcement

$$A_t = 0.0018 b \cdot h = 0.0018 (12) (6) = 0.13 \text{ in}^2/\text{ft}$$

$$\approx 4 @ 12'' \text{ o.c.} = 0.2 \text{ in}^2 > 0.13 \text{ in}^2$$

Crack Control

$$s \leq 15 \left( \frac{40000}{f_y} \right) - 2.5 c_c = 15 \left( \frac{40000}{\frac{1}{2} (60000)} \right) - 2.5 (0.75) = 13 \frac{1}{8}'' \Rightarrow s = 6'' \text{ o.c.}$$

2-38

## Determine Beam Size

Assume  $f'_c = 5000$  psi  
 Assume  $f_y = 60$  ksi

## Loads

Live Load = 80 psf (reducible)

$$L = L_o \left( 0.25 + \frac{15}{\sqrt{A_u + A_f}} \right) \geq 0.5 L_o$$

$$= 80 \left( 0.25 + \frac{15}{\sqrt{124 + 32}} \right) = 80 \cdot 0.771 = 61.68 \text{ psf}$$

$L \rightarrow 8327400$

## Dead Loads

$$\text{Slab} = \frac{6}{12} \cdot 150 = 75 \text{ psf}$$

Carpet with pad = 2 psf

Superimposed dead load = 15 psf

Total = 92 psf

$$w_u = [1.2(92) + 1.6(61.68)] \left( \frac{12 \times 13}{2} \right) = 2.72 \text{ klf}$$

$$M_u = \frac{w_u \cdot L_n^2}{8} = \frac{2.72 \cdot 30.33^2}{8} = 312.8 \text{ k}\cdot\text{ft} \cdot 1.1 = 344.1 \text{ k}\cdot\text{ft}$$

↑  
assuming  $q_{\text{clear}} = 20''$

↑  
estimate of self weight

## Estimate size:

$$b \cdot d^2 = 20 M_u, \quad b = 4/5 d$$

$$d^3 = 20(344.1) \left( \frac{5}{4} \right) = 20 \cdot 516$$

$$h = d + 2.5 \text{ in} = 23 \text{ in} \Rightarrow 24'', \quad k = 18''$$

$$b \cdot d^2 = 16 \cdot 215^2 = 83205 \text{ in}^3$$

## Compute self weight effects

$$w_{sw} = \frac{16 \cdot 24}{144} \cdot 150 = 450 \text{ plf}$$

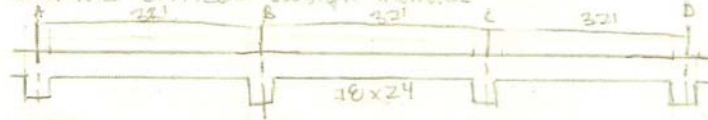
$$w_u = 2720 + 1.2(450) = 3260$$

$$M_u = \frac{3260 \cdot 30.33^2}{8} = 374.9 \text{ k}\cdot\text{ft} \quad 20 \cdot 374.9 = 7498 \text{ in}^3 < 83205 \text{ in}^3$$

$d_h$

2-39

Determine critical design moments



Use Moment Coefficient Method (ACI 318-11 Section 8.3)

- two or more spans ✓
- spans are equal ✓
- loads are uniformly distributed ✓
- L does not exceed 30 ✓
- members are prismatic ✓

Loads - (1.4k8 psf (reduced))

Dead Load

Slab = 75 psf

Beam =  $\frac{18(24-6") \cdot 150}{144 \cdot 13} = 24 \text{ psf}$ 

carpet with pad = 7 psf

superimposed dead load = 15 psf

$$w_u = [1.2(118) + 1.6(15)] \cdot 13 = 3.12 \text{ klf}$$

Critical Moments

$$M_{u,l} = w_u l^2 / 11 = \frac{3.12 \cdot 30.33^2}{11} = 260.9 \text{ k}\cdot\text{ft}$$

$$M_{u,r} = \frac{3.12 \cdot 30.33^2}{11} = 260.9 \text{ k}\cdot\text{ft}$$

$$M_u^+ = \frac{w_u l_n^2}{16} = \frac{3.12 \cdot 30.33^2}{16} = 179.4 \text{ k}\cdot\text{ft}$$

Determine Reinforcement

Check if T-beam behavior

$$b_{eff} \leq \begin{cases} 1/4 \text{ span length} = 32 \cdot 12/4 = 96" \\ b_w + 16 h_f = 18 + 16 \cdot 6 = 114" \\ b_w + 1/2 \text{ km slr spa} = 18 + (13 \cdot 12 - 20) = 156" \end{cases}$$

$$b_{eff} = 96"$$

$$M_{u,beam} = 0.85 f_c \cdot b \cdot h_f \cdot (d - h_f/2) \\ = 0.85 \cdot 4 \text{ ksi} \cdot 96" \cdot 6" \cdot (21.5 - 6/2) = 40759 \text{ kip}\cdot\text{in} > 2130.8 \text{ kip}\cdot\text{ft}$$

since  $M_u < M_{u,beam}$  then not T-beam behavior

2-40

Top-  
 $A_s \text{ req'd} = \frac{M_u}{\phi d} = \frac{260.1 \text{ k-ft}}{4 \cdot 21.6} = 3.02 \text{ in}^2 \Rightarrow 4 \# 9 \text{ bars}$

Check d

$$d = 24 - 1.5'' - 3/8'' - \frac{1.25 \phi}{2} = 21.6''$$

Check  $A_s, \text{min}$ 

$$A_s, \text{min} \geq \frac{3 \sqrt{f_c}}{f_y} b \cdot d = \frac{3 \sqrt{5000} \cdot 16 \cdot 21.6}{60000} = 1.37 \text{ in}^2$$

$$\frac{200 b \cdot d}{f_y} = \frac{200 \cdot 16 \cdot 21.6}{60000} = 1.30 \text{ in}^2$$

$$A_s = 4 \cdot (1.00) = 4.00 \text{ in}^2$$

$$A_s > A_{s, \text{min}} \checkmark$$

Check  $A_s, \text{max}$ 

$$\rho_{\text{max}} = 0.85 \beta_1 \frac{f_c'}{f_y} \frac{\epsilon_u}{\epsilon_u + 0.004} = 0.85 \cdot 0.805 \cdot \frac{0.003}{0.003 + 0.004} = 0.0243$$

$$A_{s, \text{max}} = 0.0243 \cdot 16 \cdot 21.6 = 9.45 \text{ in}^2$$

$$A_s < A_{s, \text{max}} \checkmark$$

Determine  $M_n$ Assume  $f_c \geq f_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 f_c' \cdot b} = \frac{4 \cdot 60}{0.85 \cdot 5 \cdot 16} = 3.14 \text{ in}$$

$$c = a / \beta_1 = \frac{3.14}{0.80} = 3.93 \text{ in}$$

Check  $\epsilon_s \geq \epsilon_y$ 

$$\epsilon_s = \frac{f_y}{E} (d - c) = \frac{60,000}{29,000,000} (21.6 - 3.93) = 0.0135 > 0.005$$

$$\phi M_n = \phi A_s \cdot f_y (d - a/2) = 0.9 \cdot 4 \cdot 60 \cdot (21.6 - \frac{3.14}{2}) = 360.54 \text{ k-ft}$$

$$360.54 \text{ k-ft} > 260.1 \text{ k-ft}$$

Use 4 # 9 @ top

2-41

Bottom

$$A_s \text{ req'd} = \frac{M_u}{f_y d} = \frac{179.4 \text{ k}\cdot\text{ft}}{4 \cdot 21.5} = 2.09 \text{ in}^2 \Rightarrow 3 \#9 \text{ bars}$$

check d

$$d = 24 - 1.5 - 3/8" - \frac{1.25}{2} = 21.6"$$

check  $A_s, \text{min}$ 

$$A_s, \text{min} \geq \begin{cases} \frac{3\sqrt{f_c} b \cdot d}{f_y} = \frac{3\sqrt{5000} \cdot 18 \cdot 21.6}{60000} = 1.37 \text{ in}^2 \\ \frac{200 b \cdot d}{f_y} = \frac{200 \cdot 18 \cdot 21.6}{60000} = 1.3 \text{ in}^2 \end{cases}$$

$$A_s = 3 \cdot 1.00 = 3.00 \text{ in}^2$$

$$A_s > A_s, \text{min} \checkmark$$

check  $A_s, \text{max}$ 

$$\rho_{\text{max}} = \frac{0.85 \cdot \beta_1 \cdot f_c \cdot \xi_u}{f_y \cdot \xi_u + 0.004} = \frac{0.85 \cdot 0.85 \cdot 5 \cdot 0.003}{60 \cdot 0.003 + 0.004} = 0.0243$$

$$A_s, \text{max} = 0.0243 \cdot 18 \cdot 21.6 = 9.45 \text{ in}^2$$

$$A_s < A_s, \text{max} \checkmark$$

Determine  $M_n$ assume  $f_s = f_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 \cdot f_c \cdot b} = \frac{3.0 \cdot 60}{0.85 \cdot 5 \cdot 18} = 2.35 \text{ in}$$

$$c = \frac{a}{\beta_1} = \frac{2.35}{0.8} = 2.94 \text{ in}$$

check  $\epsilon_s \geq \epsilon_y$ 

$$\epsilon_s = \frac{\xi_u}{c} (d - c) = \frac{0.003}{2.94} (21.6 - 2.94) = 0.019 > 0.005$$

$$\phi M_n = \phi A_s \cdot f_y (d - \frac{a}{2}) = 0.9 \cdot 3 \cdot 60 \cdot (21.6 - \frac{2.35}{2}) = 2757 \text{ k}\cdot\text{ft}$$

$$2757 \text{ k}\cdot\text{ft} > 179.4 \text{ k}\cdot\text{ft} \quad \text{use } 3 \#9 \text{ @ bottom}$$

2-42

Determine Shear Reinforcement

Determine Shear Strength without stirrups

$$V_c = 2\sqrt{f_c} \cdot b_w \cdot d = 2 \cdot \sqrt{5000} \cdot 18 \cdot 21.6/1000 = 54.98 \text{ k}$$

$$\phi V_n = 0.5\phi V_c = 0.5 \cdot 0.75 \cdot 54.98 \text{ k} = 20.62 \text{ k}$$

Determine shear strength required by reinforcing

$$V_u = \frac{w_u \cdot l_n}{2} = \frac{3.12 \cdot 30.33}{2} = 47.31 \text{ k}$$

$$V_s = V_u / \phi - V_c = \frac{47.31}{0.75} - 54.98 = 8.1 \text{ k}$$

$$V_{s, \text{max}} = 8\sqrt{f_c} \cdot b_w \cdot d = 8\sqrt{5000} \cdot 18 \cdot 21.6 = 219.9 \text{ k} > V_s \text{ ok}$$

Maximum spacing of shear reinforcement

$$4\sqrt{f_c} \cdot b_w \cdot d = 4 \cdot \sqrt{5000} \cdot 18 \cdot 21.6 = 110 \text{ k} > V_s$$

$$S_{\text{max}} = \min \left\{ \begin{array}{l} d/2 = 10.8'' \leftarrow \text{controls} \\ 24'' \end{array} \right.$$

Minimum shear reinforcement

$$A_{v, \text{min}} = \max \left\{ \begin{array}{l} \frac{0.75\sqrt{f_c}}{F_{yt}} \cdot b_w \cdot s = \frac{0.75\sqrt{5000} \cdot 18 \cdot 10.8}{60000} = 0.172 \text{ in}^2 \\ \frac{50 \cdot b_w \cdot s}{F_{yt}} = \frac{50 \cdot 18 \cdot 10.8}{60000} = 0.162 \text{ in}^2 \end{array} \right.$$

$$(2) \text{ legs of } \#3 = 2 \cdot 0.11 = 0.22 \text{ in}^2 > 0.172 \text{ in}^2$$

Design Shear Reinforcement

$$V_s = A_v \cdot \frac{d}{s} \cdot F_{yt} \Rightarrow s = \frac{A_v \cdot F_{yt} \cdot d}{V_s} = \frac{0.22 \cdot 60 \cdot 21.6}{6.38} = 44.7'' > s_{\text{max}}$$

Use #3 @ 10" o.c.

Minimum thickness to control deflections

$$h = \frac{l}{21} = \frac{32.12}{21} = 15.29'' < 24''$$



2-45

Determine Girder A size

assume  $f_c = 5000$  psiassume  $f_y = 60$  ksi

Loads

Live Load - 80 psf (reducible)

$$L = L_o \left( 0.25 + \frac{15}{\sqrt{K_L \cdot L_T}} \right) \geq 0.5 L_o$$

$$= 80 \left( 0.25 + \frac{15}{\sqrt{24 \cdot 32}} \right) = 80 \cdot 0.771 = 61.68 \text{ psf}$$

→ 327400

Dead Loads

Slab = 75 psf

carpet with pad = 2 psf

Superimposed dead load = 15 psf

total = 92 psf

$$w_u = [1.2(92) + 1.6(61.68)] \left( \frac{32 \cdot 32}{8} \right) = 2.72 \text{ k/ft}$$

$$M_u = \frac{w_u \cdot l_o^2}{8} = \frac{2.72 \cdot 30.33^2}{8} = 312.8 \text{ k-ft} + \frac{1}{8} = 394.1 \text{ k-ft}$$

↑ assuming column 20"      ↑ estimate of self weight

Estimate size

$$b \cdot d^3 = 20 M_u; \quad b = 20 \text{ in}$$

$$20 = \frac{20 \cdot 394.1}{d^3} \quad d = 18.16 \text{ in} \rightarrow h = 24 \text{ in min}$$

d = 21.5 in

$$b \cdot d^2 = 20 \cdot 21.5^2 = 9245 \text{ in}^2$$

Compute self weight effects

$$w_{sw} = \frac{20 \cdot 24}{144} \cdot 150 = 500 \text{ plf}$$

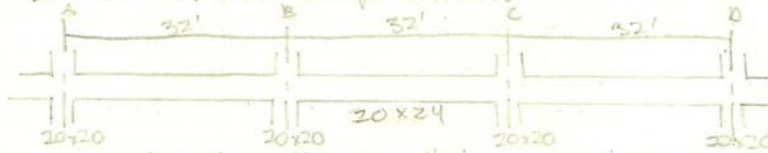
$$w_u = 2.720 + 1.2(500) = 3.320 \text{ plf}$$

$$M_u = \frac{3.32 \cdot 30.33^2}{8} = 381.76 \text{ k-ft} \quad 20 \cdot 381.76 = 7635.2 \text{ k-in}^3$$

OK

2-44

Determine critical design moments



Use Moment Coefficient Method

- two or more spans ✓
- spans are equal ✓
- Loads are uniformly distributed ✓
- $l$  does not exceed  $3l$  ✓
- Members are prismatic ✓

Loads - 6.16 k/psf

Dead Load

Slab = 75 psf

$$\text{girder} = \frac{20 \cdot (24 - 6) \cdot 150}{144 \cdot 12} = 29 \text{ psf}$$

carpet with pad = 2 psf

Superimposed dead load = 15 psf

$$w_u [1.2(121) + 1.6(61.68)] \cdot 13 = 3.17 \text{ k/ft}$$

Critical Moments

$$M_u^-, l = \frac{w_u l^2}{16} = \frac{3.17 \cdot 30.33^2}{16} = 265.1 \text{ k}\cdot\text{ft}$$

$$M_u^-, r = \frac{w_u l^2}{16} = \frac{3.17 \cdot 30.33^2}{16} = 265.1 \text{ k}\cdot\text{ft}$$

$$M_u^+ = \frac{w_u l^2}{16} = \frac{3.17 \cdot 30.33^2}{16} = 182.3 \text{ k}\cdot\text{ft}$$

Determine Reinforcement

Check if T-beam behavior

$$b_{eff} \leq \begin{cases} 1/4 \text{ span length} = 32 \cdot 12/4 = 96'' \\ b_w + 16h_f = 20 + 16 \cdot 6 = 116'' \\ b_w + 2 \cdot 1/2 \text{ bm. in span} = 20 + (13 \cdot 12 - 18) = 158'' \end{cases}$$

$$b_{eff} = 96''$$

$$M_u^+ - b_n = 0.85 \cdot f_c \cdot b \cdot h_f \cdot (d - h_f/2) = 0.9 \cdot 0.85 \cdot 5 \cdot 96 \cdot 6 \cdot (21.5 - 6/2) = 40,759 \text{ k}\cdot\text{in} > 3130.8 \text{ k}\cdot\text{in}$$

Since  $M_u^+ < M_u^+ - b_n$  then no T-beam behavior

2-45

Top-

$$A_s \text{ req'd} = \frac{M_u}{\phi d} = \frac{265.1}{4 \cdot 21.6} = 3.08 \text{ in}^2 \rightarrow 4 \#9 \text{ bars}$$

Check d

$$d = 24 - 1.5'' - 3/8'' - \frac{1.5 \cdot 9}{16} = 21.6''$$

check  $A_{s, \min}$ 

$$A_{s, \min} \geq \begin{cases} \frac{3\sqrt{f_c}}{f_y} b \cdot d = \frac{3\sqrt{5000} \cdot 20 \cdot 21.6}{60000} = 1.53 \text{ in}^2 \\ \frac{200 b \cdot d}{f_y} = \frac{200 \cdot 20 \cdot 21.6}{60000} = 1.44 \text{ in}^2 \end{cases}$$

$$A_s = 4 \cdot 1.00 = 4.00 \text{ in}^2$$

$$A_s > A_{s, \min} \checkmark$$

Check  $A_{s, \max}$ 

$$\rho_{\max} = 0.85 \beta_1 \frac{f_c}{f_y} \frac{\epsilon_u}{\epsilon_u + 0.001} = 0.85 \cdot 0.80 \frac{5}{60} \cdot \frac{0.003}{0.003 + 0.001} = 0.0243$$

$$A_{s, \max} = 0.0243 \cdot 20 \cdot 21.6 = 10.5 \text{ in}^2$$

$$A_s < A_{s, \max} \checkmark$$

Determine  $M_n$ assume  $f_s = f_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 f_c \cdot b} = \frac{4.0 \cdot 60}{0.85 \cdot 5 \cdot 20} = 2.82 \text{ in}$$

$$c = a / \beta_1 = \frac{2.82}{0.8} = 3.53 \text{ in}$$

Check  $\epsilon_s > \epsilon_y$ 

$$\epsilon_s = \frac{\epsilon_u (d - c)}{c} = \frac{0.003 (21.6 - 3.53)}{3.53} = 0.015 > 0.005$$

$$\phi M_n = \phi A_s f_y (d - a/2) = 0.9 \cdot 4 \cdot 60 (21.6 - \frac{2.82}{2}) = 363.42 \text{ k}\cdot\text{ft}$$

$$363.42 \text{ k}\cdot\text{ft} > 265.1 \text{ k}\cdot\text{ft}$$

use 4#9 @ top

Bottom

$$A_s \text{ req'd} = \frac{M_u}{\phi_f} = \frac{182.3}{4 \cdot 21.5} = 2.12 \text{ in}^2 \rightarrow 3 \#9 \text{ bars}$$

check d

$$d = 24 - 1.5 - 3/8 - \frac{6 \cdot 3/8}{2} = 21.6''$$

Check  $A_{s, \min}$ 

$$A_{s, \min} \geq \frac{3 \sqrt{f_c'} \cdot b \cdot d}{f_y} = \frac{3 \sqrt{5000} \cdot 20 \cdot 21.6}{60000} = 1.53 \text{ in}^2$$

$$\frac{200 b \cdot d}{f_y} = \frac{200 \cdot 20 \cdot 21.6}{60000} = 1.44 \text{ in}^2$$

$$A_s = 3 \cdot 1.00 = 3.00 \text{ in}^2$$

$$A_s > A_{s, \min}$$

Check  $A_{s, \max}$ 

$$\rho_{\max} = 0.85 \cdot \beta_1 \cdot \frac{f_c'}{f_y} \cdot \frac{\epsilon_u}{\epsilon_u + \epsilon_s} = 0.85 \cdot 0.85 \cdot \frac{5}{60} \cdot \frac{0.003}{0.003 + 0.004} = 0.0243$$

$$A_{s, \max} = 0.0243 \cdot 20 \cdot 21.6 = 10.5 \text{ in}^2$$

$$A_s < A_{s, \max} \checkmark$$

Determine  $M_n$ assume  $f_s > f_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 \cdot f_c' \cdot b} = \frac{3 \cdot 60}{0.85 \cdot 5 \cdot 20} = 2.12 \text{ in}$$

$$c = a / \beta_1 = \frac{2.12}{0.85} = 2.49 \text{ in}$$

Check  $\epsilon_s > \epsilon_y$ 

$$\epsilon_s = \frac{\epsilon_u}{c} (d - c) = \frac{0.003}{2.49} (21.6 - 2.49) = 0.0213 > 0.002$$

$$\phi M_n = \phi A_s \cdot f_y \left( d - \frac{a}{2} \right) = 0.9 \cdot 3 \cdot 60 \cdot \left( 21.6 - \frac{2.12}{2} \right) = 273.7 \text{ k-ft}$$

$$273.7 \text{ k-ft} > 182.5 \text{ k-ft}$$

use 3 #9 @ bottom.

2-47

Determine Shear Reinforcement

Determine shear strength without stirrups

$$V_c = 2\sqrt{F_c} b_w d = 2 \cdot \sqrt{5000} \cdot 20 \cdot 21.6 / 1000 = 61.1 \text{ k}$$

$$\phi V_n = 0.5\phi V_c = 0.5 \cdot 0.75 \cdot 61.1 = 22.9 \text{ k}$$

Determine shear strength required by reinforcing

$$V_u = \frac{w_u l_n}{2} = \frac{3.17 \cdot 30.53}{2} = 48.07 \text{ k}$$

$$V_s = V_u / \phi - V_c = \frac{48.07}{0.75} - 61.1 = 3 \text{ k}$$

$$V_{s, \max} = 8\sqrt{F_c} b_w d = 8\sqrt{5000} \cdot 20 \cdot 21.6 = 244.4 \text{ k} > V_s$$

Maximum spacing of shear reinforcement

$$4\sqrt{F_c} \cdot b_w \cdot d = 4 \cdot \sqrt{5000} \cdot 20 \cdot 21.6 = 122.19 \text{ k} > V_s$$

$$s_{\max} = \min \left\{ \begin{array}{l} d/2 = 10.8'' \leftarrow \text{controls} \\ 24'' \end{array} \right.$$

Minimum shear Reinforcement

$$A_{v, \min} = \max \left\{ \begin{array}{l} \frac{0.75\sqrt{F_c} \cdot b_w \cdot s}{f_{yt}} = \frac{0.75 \cdot \sqrt{5000} \cdot 20 \cdot 10.8}{60000} = 0.19 \text{ in}^2 \\ \frac{50 b_w \cdot s}{f_{yt}} = \frac{50 \cdot 20 \cdot 10.8}{60000} = 0.18 \text{ in}^2 \end{array} \right.$$

$$2 \text{ legs of } \#3 = 2 \cdot 0.11 = 0.22 \text{ in}^2 > 0.19 \text{ in}^2$$

Design Shear Reinforcement

$$V_s = A_v \cdot \frac{d}{s} \cdot f_{yt} \Rightarrow s = \frac{A_v \cdot f_{yt} \cdot d}{V_s} = \frac{0.22 \cdot 60 \cdot 21.6}{3} = 95 \text{ in} > s_{\max}$$

Use #3 @ 10" o.c.

Minimum thickness to control deflections

$$h = l/21 = \frac{32 \cdot 12}{21} = 18.29 > 24''$$

2-48

Determine Winder B size

Assume  $f'_c = 5000$  psi  
Assume  $f_y = 60$  ksi

Loads

Live Load - 80 psf (reducible)

$$L = L_0 \left( 0.25 + \frac{15}{\sqrt{K_{nc} \cdot A_t}} \right) \geq 0.5 L_0$$

$$= 80 \left( 0.25 + \frac{15}{\sqrt{32 \cdot 2 \cdot 26}} \right) = 80 \cdot 0.563 = 45.04 \text{ psf}$$

$\rightarrow 1664 \approx 100$

Dead Loads

Slab - 75 psf  
Carpet with pad - 7 psf  
Superimposed dead load - 15 psf  
beam self weight - 26 psf

Total - 118

$$w_u = 1.2(118) + 1.6(45.04) = 213.67 \text{ psf}$$

$$P_u = 213.67(13)(32) = 90.9 \text{ k}$$

$$M_u = \frac{P_u l_n}{4} = \frac{90.9(25.5)}{4} = 580.8 \text{ k}\cdot\text{ft} + 1.2 = 649.8 \text{ k}\cdot\text{ft}$$

(estimate of self weight)

Estimate size

$$b \cdot d^2 = 20 M_u, \quad b = 20 \text{ in}$$

$$20 = 20 \cdot \frac{649}{d^2} \quad d = 25.5 \text{ in} \rightarrow h = 28 \text{ in}$$

d = 25.5 in

$$b \cdot d^2 = 20 \cdot 25.5^2 = 13005 \text{ in}^3$$

Compute self weight effects

$$w_{sw} = \frac{20 \cdot 28}{144} \cdot 150 = 583 \text{ plf}$$

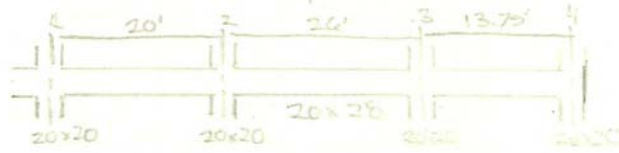
$$P_u = 90.9 \text{ k} + 0.563 \cdot 26 = 104.1 \text{ k}$$

$$M_u = \frac{104.1(25.5)}{4} = 663.3 \text{ k}\cdot\text{ft} + 20 \cdot 663.3 = 12666 \text{ in}^3$$

13005 in<sup>3</sup>

2-49

Determine critical design moments



Use moment coefficient method

Loads

Live Load - 45.04 psf

Dead Load

slab - 7.5 psf

beam - 2.6 psf

girder -  $\frac{20 \cdot (28 - 6) \cdot 150}{144 \cdot 32} = 14.3$  psf

carpet with pad - 2 psf

Superimposed dead load - 15 psf

$$w_u = [1.2(15 + 2 + 3) + 1.6(45.04)] \cdot 32 = 7.39 \text{ k/ft}$$

Critical Moments

$$M_{u,r,l} = \frac{w_u l_n^2}{11} = \frac{7.39 \cdot 24.33^2}{11} = 397.7 \text{ k}\cdot\text{ft}$$

$$M_{u,r,r} = \frac{w_u l_n^2}{11} = \frac{7.39 \cdot 24.33^2}{11} = 397.7 \text{ k}\cdot\text{ft}$$

$$M_u^+ = \frac{w_u l_n^2}{16} = \frac{7.39 \cdot 24.33^2}{16} = 273.41 \text{ k}\cdot\text{ft}$$

Determine Reinforcement

Top

$$A_s \text{ req'd} = \frac{M_u}{\phi d} = \frac{397.7}{4 \cdot 25.5} = 3.89 \text{ in}^2 \rightarrow 4\#9 \text{ bars}$$

check d

$$d = 28 - 1.5 - \frac{3}{8} - \frac{4 \cdot 3}{2} = 25.6 \text{ in}$$

2-50

Check  $A_s, \min$ 

$$A_{s, \min} \geq \begin{cases} \frac{3\sqrt{f_c} \cdot b \cdot d}{f_y} = \frac{3\sqrt{5000} \cdot 20 \cdot 25.6}{60000} = 1.81 \text{ in}^2 \\ \frac{200 \cdot b \cdot d}{f_y} = \frac{200 \cdot 20 \cdot 25.6}{60,000} = 1.71 \text{ in}^2 \end{cases}$$

$$A_s = 4 \cdot 1.00 = 4 \text{ in}^2$$

$$A_s > A_{s, \min} \checkmark$$

Check  $A_s, \max$ 

$$\rho_{\max} = 0.85 \cdot \beta_1 \cdot \frac{f_c}{f_y} \cdot \frac{\epsilon_u}{\epsilon_u + 0.004} = 0.85 \cdot 0.85 \cdot \frac{5}{60} \cdot \frac{0.00375}{0.00375 + 0.004} = 0.0243$$

$$A_{s, \max} = 0.0243 \cdot 20 \cdot 25.6 = 12.4 \text{ in}^2$$

$$A_s < A_{s, \max} \checkmark$$

Determine  $M_n$ assume  $f_s = f_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 \cdot f_c \cdot b} = \frac{4 \cdot 60}{0.85 \cdot 5 \cdot 20} = 2.82 \text{ in}$$

$$c = a / \beta_1 = \frac{2.82}{0.85} = 3.53 \text{ in}$$

Check  $\epsilon_s > \epsilon_y$ 

$$\epsilon_s = \frac{\epsilon_u}{c} (d - c) = \frac{0.00375}{3.53} (25.6 - 3.53) = 0.0188 > 0.005$$

$$\phi M_n = \phi A_s f_y (d - a/2) = 0.9 \cdot 4 \cdot 60 \cdot (25.6 - \frac{2.82}{2}) = 435.4 \text{ k}\cdot\text{ft}$$

$$435.4 \text{ k}\cdot\text{ft} > 397.7 \text{ k}\cdot\text{ft}$$

use 4 # 9 @ top



2-51

Bottom

$$A_s \text{ req'd} = \frac{M_u}{4d} = \frac{273.41}{4 \cdot 21.5} = 3.12 \text{ in}^2 \rightarrow 4 \# 9 \text{ bars}$$

Check  $A_{s, \min}$ 

$$A_{s, \min} \geq \begin{cases} \frac{3\sqrt{f'_c} \cdot b \cdot d}{f_y} = \frac{3\sqrt{5000} \cdot 20 \cdot 25.6}{60,000} = 1.53 \text{ in}^2 \\ \frac{200 \cdot b \cdot d}{f_y} = \frac{200 \cdot 20 \cdot 25.6}{60,000} = 1.3 \text{ in}^2 \end{cases}$$

$$A_s = 4 \cdot 1.00 = 4.00 \text{ in}^2$$

$$A_s > A_{s, \min}$$

Check  $A_{s, \max}$ 

$$\rho_{\max} = 0.85 \cdot \beta_1 \cdot \frac{f'_c}{f_y} \cdot \frac{\epsilon_u}{\epsilon_u + 0.001} = 0.85 \cdot 0.8 \cdot \frac{5}{60} \cdot \frac{0.003}{0.003 + 0.001} = 0.0243$$

$$A_{s, \max} = 0.0243 \cdot 20 \cdot 25.6 = 12.44 \text{ in}^2$$

$$A_s < A_{s, \max} \checkmark$$

Determine  $M_n$ assume  $f_s > f_y$ 

$$a = \frac{A_s \cdot f_y}{0.85 \cdot f'_c \cdot b} = \frac{4.0 \cdot 60}{0.85 \cdot 5 \cdot 20} = 2.82 \text{ in}$$

$$c = a/\beta_1 = \frac{2.82}{0.8} = 3.53 \text{ in}$$

Check  $\epsilon_s > \epsilon_y$ 

$$\epsilon_s = \frac{\epsilon_u (d - c)}{c} = \frac{0.003 (25.6 - 3.53)}{3.53} = 0.0185 > 0.003$$

$$\phi M_n = \phi A_s f_y (d - a/2) = 0.9 (4)(60) (25.6 - \frac{2.82}{2}) = 435.4 \text{ k}\cdot\text{ft}$$

$$435.4 \text{ k}\cdot\text{ft} > 273.41 \text{ k}\cdot\text{ft}$$

Use 4 # 9 @ bottom

2-52

Determine shear Reinforcement

Determine shear strength without stirrups

$$V_c = 2 \sqrt{f_c'} b_w \cdot d = 2 \sqrt{5000} \cdot 20 \cdot 25.6 / 1000 = 72.4 \text{ k}$$

$$\phi V_n = 0.5 \phi V_c = 0.5 \cdot 0.75 \cdot 72.4 \text{ k} = 37.2 \text{ k}$$

Determine shear strength required by reinforcing

$$V_u = \frac{w_u b h}{2} = \frac{7.39 \cdot 24 \cdot 33}{2} = 89.9 \text{ k}$$

$$V_s = V_u / \phi - V_c = \frac{89.9}{0.75} - 72.4 = 47.5 \text{ k}$$

$$V_{s, \text{max}} = 8 \sqrt{f_c'} b_w \cdot d = 8 \sqrt{5000} \cdot 20 \cdot 25.6 = 259.6 \text{ k} \quad \phi V_s >$$

Maximum Spacing of Shear Reinforcement

$$4 \sqrt{f_c'} \cdot b_w \cdot d = 4 \sqrt{5000} \cdot 20 \cdot 25.6 = 149.8 \text{ k} > V_s \text{ ok}$$

$$s_{\text{max}} = \min \left\{ \frac{d}{2} = 12.8'' \neq \text{controls} \right. \\ \left. 24'' \right.$$

Minimum shear reinforcement

$$A_{v, \text{min}} = \max \left\{ \frac{0.75 \sqrt{f_c'} \cdot b_w \cdot s}{f_{yt}} = \frac{0.75 \cdot \sqrt{5000} \cdot 20 \cdot 12.8}{60000} = 0.226 \text{ in}^2 \right.$$

$$\left. \frac{50 b_w \cdot s}{f_{yt}} = \frac{50 \cdot 20 \cdot 12.8}{60000} = 0.213 \text{ in}^2 \right.$$

$$(3) \text{ legs of } \# 3 = 3 \cdot 0.11 = 0.33 \text{ in}^2 > 0.226 \text{ in}^2$$

Design Shear Reinforcement

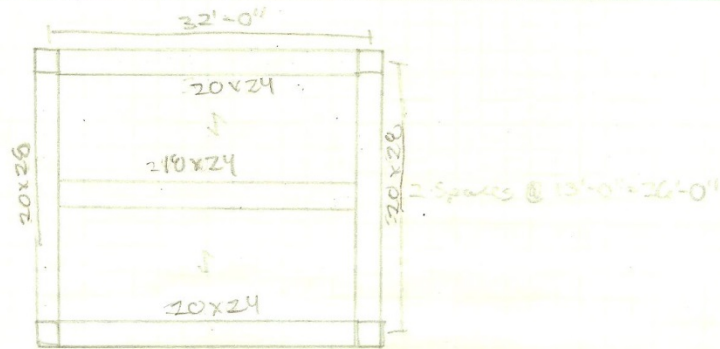
$$V_s = A_v \cdot \frac{d}{s} f_{yt} \Rightarrow s = \frac{A_v \cdot f_{yt} \cdot d}{V_s} = \frac{0.33 \cdot 60 \cdot 25.6}{47.5} = 10.67 \text{ in}$$

Use  $\# 3$  legs @ 10" o.c.

Minimum Thickness to control deflections

$$h = \frac{L}{21} = \frac{26 \cdot 12}{21} = 14.85'' < 28''$$

2-53



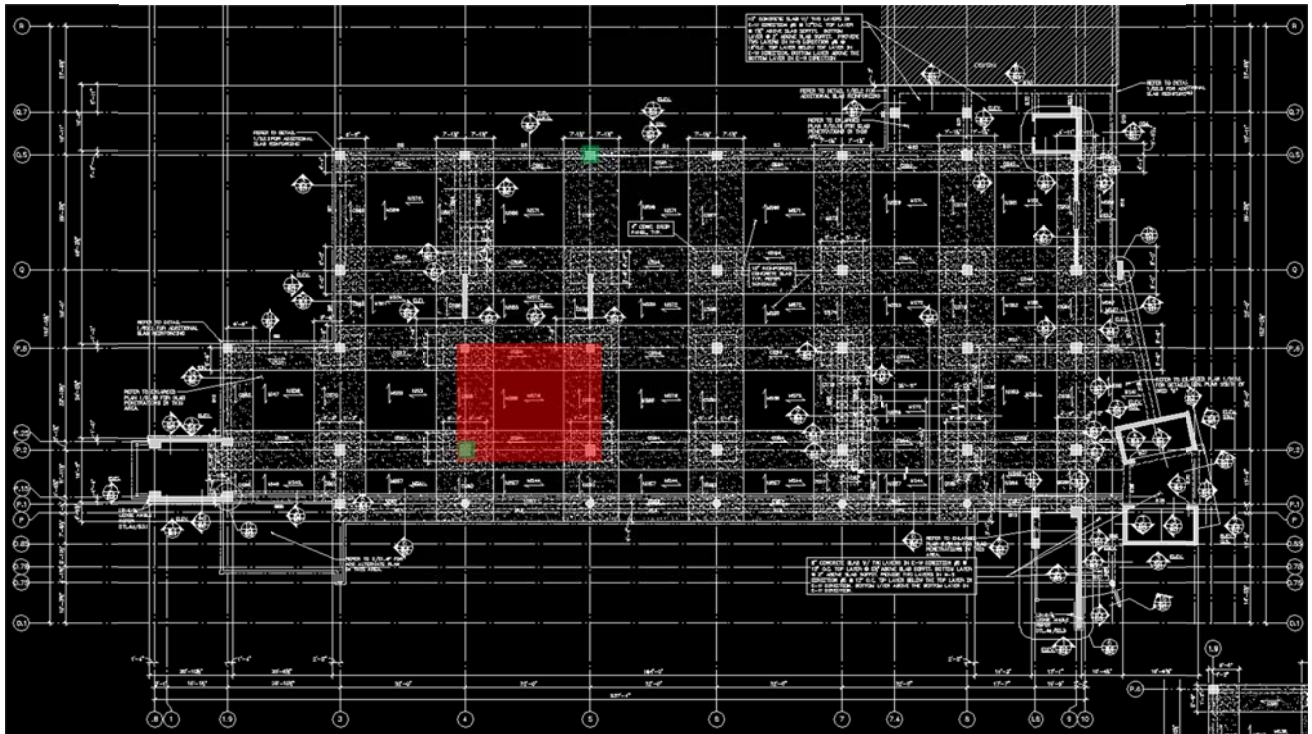
Beam Schedule

Member	f'c	Top flexural Reinforcement	Bottom flexural Reinforcement	Shear Reinforcement
10x24	5000	4#9	3#9	(2) legs #3 @ 10" o.c.
20x24	5000	4#9	3#9	(2) legs #3 @ 10" o.c.
20x28	5000	4#9	4#9	(2) legs #3 @ 10" o.c.

Member	f'c	Longitudinal Reinforcement	Transverse Reinforcement	Thickness
slab	5000	#4 bar @ 12 in o.c.	#4 bar @ 12 in o.c.	6"

AMND

## Appendix A - Floor Plan with Typical Bay and Columns



Typical bay is outlined in red. Interior and exterior columns are outlined in green.